



TECHNICAL NOTE

D-822

TABLES OF AIRSPEED, ALTITUDE, AND MACH NUMBER
BASED ON LATEST INTERNATIONAL VALUES
FOR ATMOSPHERIC PROPERTIES
AND PHYSICAL CONSTANTS

By Sadie P. Livingston and William Gracey

Langley Research Center
Langley Field, Va.

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
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N O T I C E

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ERRATA

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Page 14: In column 1 of table I, the nine values of calibrated
airspeed, V_c , mph, following the value 1,000 should be corrected
as follows:

Value shown	Correct value
1,100	1,010
1,200	1,020
1,300	1,030
1,400	1,040
1,500	1,050
1,600	1,060
1,700	1,070
1,800	1,080
1,900	1,090

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SUMMARY

Previously published pressure-airspeed tables have been revised to account for the recently established value for the nautical mile and to incorporate the values of the physical constants and atmospheric properties adopted for the International Civil Aviation Organization (ICAO) standard atmosphere. In the revised tables, values of impact pressure are given for airspeeds up to 1,000 knots and 1,100 miles per hour. Pressure-altitude tables, extracted from the ICAO standard atmosphere, are given for altitudes up to 100,000 geopotential feet. For the determination of Mach number, the ratio of impact pressure to free-stream static pressure is given for Mach numbers from 0.10 to 5.00. Abbreviated tables of true airspeed are given for calibrated airspeeds from 100 to 1,000 knots and for altitudes from 0 to 100,000 geopotential feet.

INTRODUCTION

Pressure-airspeed and pressure-altitude tables are needed for the calibration of airspeed meters and altimeters and for the conversion of flight measurements of airspeed and altitude to other related parameters - Mach number, true airspeed, equivalent airspeed, etc. Equations relating these parameters and the procedures for determining them are given in reference 1.

Numerous airspeed and altitude tables have been developed through the years (refs. 1 to 11). The airspeed tables of references 1 to 3 have undergone minor revisions as new values for some of the physical constants and atmospheric properties required for the computations have been introduced. The altitude tables of references 4 to 11, on the other hand, have been revised extensively as new information concerning atmospheric properties (particularly at the higher altitudes)

has become available and because of the introduction of a new unit of height - geopotential feet, a unit that takes account of the decrease of the gravity constant with height.

Concerted efforts have been made in recent years to adopt, on an international basis, the latest accepted values of the atmospheric properties and a consistent system of physical constants. The result of these efforts culminated in the establishment of the ICAO standard atmosphere in 1954 (ref. 8) and an extension of this atmosphere to higher altitudes (ref. 10) in 1958. It may be noted that the pressure-altitude tables of references 7 and 8 are in agreement for altitudes up to 65,800 geopotential feet (the limiting altitude in ref. 8) and the tables of references 9, 10, and 11 are in agreement for altitudes up to at least 100,000 geopotential feet (the upper limit of the altitude range of interest at the present time for the calibration of aircraft altimeters).

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With regard to the airspeed tables, a need has arisen for a reevaluation of the pressure-airspeed relation to account for the recently revised value of the nautical mile, with consequent changes of speed in units of knots. At the same time, it appeared desirable, in the interest of standardization, to incorporate in the pressure-airspeed tables the values for the atmospheric properties and physical constants adopted for the ICAO standard atmosphere.

Accordingly, the pressure-airspeed tables of reference 1 have been revised to account for the new value of the nautical mile and the latest accepted values for the pertinent physical and atmospheric-property constants. At the same time, the airspeed tables of reference 1 have been extended to higher speeds to conform to the maximum range of present-day airspeed indicators. In addition, the pressure-altitude tables of reference 1 have been replaced with those of the ICAO standard atmosphere and the Mach number tables have been extended to a Mach number of 5.00.

SYMBOLS

ρ	mass density of the ambient air
p	free-stream static pressure
p_t	free-stream total pressure
q	dynamic pressure, $\frac{1}{2} \rho V^2$
q_c	impact pressure, $p_t - p$

f	compressibility factor (defined in ref. 1)
V	free-stream velocity or true airspeed, $\sqrt{\frac{2q}{\rho}}$
V _c	calibrated airspeed (the indication of a differential-pressure airspeed indicator, calibrated to read true airspeed under standard sea-level conditions and corrected for instrument errors and errors of pitot-static installation), $f_0 \sqrt{\frac{2q_c}{\rho_0}}$
γ	ratio of specific heat of air at constant pressure to specific heat at constant volume
a	speed of sound in ambient air, $\sqrt{\frac{\gamma p}{\rho}}$
M	free-stream Mach number, V/a
Z	geometric altitude
H	geopotential altitude
T _M	molecular-scale temperature (defined in ref. 10)
W _M	molecular weight of air
R	universal gas constant
g	acceleration of gravity
G	proportionality factor depending on units of H (defined in ref. 10)
r	effective radius of earth at latitude 45°32'33"
t	temperature
T	absolute temperature
η	recovery factor of air-temperature sensor
Subscripts:	
o	sea-level value
l	local value
m	measured value

PHYSICAL CONSTANTS AND SEA-LEVEL ATMOSPHERIC PROPERTIES

Since the only tables in reference 1 which have been recomputed for the present report are those relating airspeed to pressure and the speed of sound to temperature, only those physical constants and sea-level atmospheric properties applying to these relations are given herein. For the values of the other constants and atmospheric properties used in the computation of the pressure-altitude tables, consult references 9 to 11.

The values of the physical constants and sea-level atmospheric properties used for the computation of the tables given in this report are as follows:

$$p_0 = 29.92126 \text{ inches of mercury or } 2116.217 \text{ pounds per square foot}$$

$$\rho_0 = 0.00237692 \text{ slug per cubic foot}$$

$$T = 518.688^\circ \text{ R or } 288.16^\circ \text{ K}$$

$$\gamma = 1.4 \text{ (exact)}$$

$$a_0 = 761.2116 \text{ miles per hour or } 661.4748 \text{ knots}$$

The values of p_0 , ρ_0 , T , and γ are the latest accepted values for these quantities as reported in references 9 to 11. The values of a_0 have been derived from the value of 340.29205 meters per second given in references 9 to 11 and from the following relations:

$$1 \text{ foot} = 0.3048 \text{ meter}$$

$$1 \text{ statute mile} = 5,280 \text{ feet}$$

$$1 \text{ nautical mile} = 6076.11549 \text{ feet}$$

As discussed in reference 12, the relation between the meter and the nautical mile (1 nautical mile = 1,852 meters) was adopted by the U.S. Departments of Commerce and Defense on July 1, 1954; the relation between the foot and the meter was announced by the National Bureau of Standards on July 1, 1959. The value of the nautical mile given herein is based on these two relationships.

In this report the values of airspeed are given for pressures expressed in units of inches of mercury and pounds per square foot. Conversion factors for expressing the tabulated pressures in other units are given in the following table (from ref. 13):

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Conversion Factors for Various Pressure Units Equivalent for Unit Value in First Column

Pressure unit value	mm mercury 0° C	in. mercury 0° C	gm/cm ²	lb/in. ²	lb/ft ²	cm water 20° C	in. water 20° C
1 mm mercury	1	0.03937	1.3595	0.019337	2.7845	1.3609	0.53577
1 in. mercury	25.400	1	34.532	0.49116	70.727	34.566	13.609
1 gm/cm ²	0.73556	0.028959	1	0.014223	2.0482	1.0010	0.39409
1 lb/in. ²	51.715	2.0360	70.307	1	144	70.376	27.707
1 lb/ft ²	0.35913	0.014139	0.48824	0.0069444	1	0.48872	0.19241
1 cm water 20° C	0.73424	0.028907	0.99821	0.014198	2.0445	1	0.3937
1 in. water 20° C	1.8650	0.073424	2.5355	0.036063	5.1930	2.5400	1

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AIRSPEED TABLES

For the conventional method of determining the speed of aircraft from measurements of free-stream total and static pressures, true airspeed is related to these pressures in the subsonic speed range by Bernoulli's formula for the total pressure in compressible flow,

$$P_t = P \left(1 + \frac{\gamma - 1}{2\gamma} \frac{\rho v^2}{P} \right)^{\frac{\gamma}{\gamma - 1}} \quad (1)$$

and in the supersonic range, where a normal shock occurs ahead of the pitot tube, by the relation for the total pressure behind a normal shock,

$$P_t = \frac{1 + \gamma}{2\gamma} \rho v^2 \left[\frac{\frac{(\gamma + 1)^2}{\gamma} \frac{\rho}{P} v^2}{\frac{4\rho v^2}{P} - 2(\gamma - 1)} \right]^{\frac{1}{\gamma - 1}} \quad (2)$$

Since the impact pressure is defined as the difference between the total and static pressures

$$q_c = P_t - P \quad (3)$$

and the speed of sound in ambient air may be expressed as

$$a = \sqrt{\gamma \frac{p}{\rho}} \quad (4)$$

the total-pressure—airspeed relation in equations (1) and (2) may be expressed in terms of q_c and V as follows:

$$q_c = p \left[\left(1 + \frac{\gamma - 1}{2\gamma} \frac{\rho V^2}{p} \right)^{\frac{\gamma}{\gamma - 1}} - 1 \right] \quad (V \leq a) \quad (5)$$

and

$$q_c = \frac{1 + \gamma}{2} \left(\frac{V}{a} \right)^2 p \left[\frac{(\gamma + 1)^2}{4\gamma - 2(\gamma - 1) \left(\frac{a}{V} \right)^2} \right]^{\frac{1}{\gamma - 1}} - p \quad (V \geq a) \quad (6)$$

Aircraft airspeed indicators are calibrated according to equations (5) and (6) for standard sea-level conditions - that is,

$$q_c = p_o \left[\left(1 + \frac{\gamma - 1}{2\gamma} \frac{\rho_o V_c^2}{p_o} \right)^{\frac{\gamma}{\gamma - 1}} - 1 \right] \quad (V_c \leq a_o) \quad (7)$$

and

$$q_c = \frac{1 + \gamma}{2} \left(\frac{V_c}{a_o} \right)^2 p_o \left[\frac{(\gamma + 1)^2}{4\gamma - 2(\gamma - 1) \left(\frac{a_o}{V_c} \right)^2} \right]^{\frac{1}{\gamma - 1}} - p_o \quad (V_c \geq a_o) \quad (8)$$

where the subscript o denotes standard sea-level conditions and V_c is the calibrated airspeed. For standard sea-level conditions, therefore, calibrated airspeed is equal to true airspeed. For other values of pressure and density, the true airspeed and the calibrated airspeed may be related by the following equation:

$$V = V_c \frac{p}{p_o} \sqrt{\frac{\rho_o}{\rho}} \quad (9)$$

where f and f_0 are compressibility factors, the values of which may be found in reference 1. True airspeed can also be determined from calibrated airspeed by another method which is described in a subsequent section.

Values of impact pressure q_c (calculated from eqs. (7) and (8)) in units of inches of mercury and pounds per square foot are given in tables I to IV for calibrated airspeeds V_c from 0 to 1,000 knots and from 0 to 1,100 miles per hour.

ALTITUDE TABLES

Pressure-altitude relations are derived from the barometric equation which may be expressed in the following form:

$$d \log_e p = - \frac{g W_M}{RT} dz \quad (10)$$

where Z denotes height in geometric measure.

In the computation of the altitude tables of references 4 to 6, the acceleration of gravity g and the molecular weight of air W_M were assumed to remain constant at their sea-level values. For the pressure-altitude tables in references 7 to 11, the decrease in acceleration of gravity with height was taken into account by the formation of a new parameter - geopotential altitude H (the combination of geometric height and acceleration of gravity). For altitudes above 295,000 feet the decrease in the molecular weight of air was taken into account in references 9 to 11 by the formation of another parameter - molecular-scale temperature T_M (the combination of molecular weight and temperature). In accordance with these new parameters the barometric equation assumes the following form:

$$d \log_e p = - \frac{G W_{M,0}}{RT_M} dH \quad (11)$$

where G is a proportionality factor depending on the units of geopotential altitude H . It may be noted that in references 9 to 11 the symbol M was assigned to molecular weight; in the present paper W_M is used to avoid confusion with the symbol M for Mach number.

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The relation between geopotential and geometric altitudes, based on the inverse-square-law variation of acceleration of gravity, is given by the equation:

$$H = \frac{g_0}{G} \frac{rZ}{r+Z} \quad (12)$$

where $r = 6,356,766$ geometric meters, the effective radius of the earth at latitude $45^{\circ}32'33''$ (refs. 9 to 11).

Since the current trend in the aviation industry is to adopt geopotential measure for the calibration of altimeters, the pressure-altitude tables in the present report are given in terms of geopotential feet only.

Tables V and VI give values of static pressure p in inches of mercury and pounds per square foot, respectively, for values of pressure altitude up to 100,000 geopotential feet. The values of p for pressure altitudes from -1,000 to 65,800 geopotential feet were taken directly from reference 8 and those for pressure altitudes from 70,000 to 100,000 geopotential feet were taken directly from reference 10.

MACH NUMBER TABLES

Since Mach number M is related to true airspeed V by the equation

$$V = M \sqrt{\gamma \frac{p}{\rho}} \quad (13)$$

M may be related to p_t by substitution into equations (1) and (2). With the additional substitution of $q_c + p$ for p_t , these equations may be expressed in terms of q_c and M as follows:

$$q_c = p \left[\left(1 + \frac{\gamma - 1}{2} M^2 \right)^{\frac{\gamma}{\gamma - 1}} - 1 \right] \quad (M \leq 1) \quad (14)$$

and

$$q_c = \frac{1 + \gamma M^2}{2} p \left[\frac{(1 + \gamma)^2 M^2}{4\gamma M^2 - 2(\gamma - 1)} \right]^{\frac{1}{\gamma - 1}} - p \quad (M \geq 1) \quad (15)$$

Since, as shown by these equations, Mach number is a function of both impact and static pressures, M may be expressed as a function of the ratio q_c/p in the following forms:

$$\frac{q_c}{p} = \left(1 + \frac{\gamma - 1}{2} M^2\right)^{\frac{\gamma}{\gamma - 1}} - 1 \quad (M \leq 1) \quad (16)$$

and

$$\frac{q_c}{p} = \frac{1 + \gamma}{2} M^2 \left[\frac{(1 + \gamma)^2 M^2}{4\gamma M^2 - 2(\gamma - 1)} \right]^{\frac{1}{\gamma - 1}} - 1 \quad (M \geq 1) \quad (17)$$

The application of equation (17), like that for equations (6) and (8), requires that the pitot tube always operate behind a normal shock at supersonic speeds.

Table VII gives the values of the ratio of impact pressure to static pressure q_c/p for values of Mach number M from 0.10 to 5.00; these values were extracted from reference 3.

TRUE AIRSPEED TABLES

From the definition of Mach number - that is, the ratio of true airspeed to the speed of sound in the ambient air - true airspeed can be calculated from the following equation:

$$V = Ma \quad (18)$$

Since the speed of sound is a function of the absolute temperature of the ambient air in accordance with the equation

$$a = \sqrt{\gamma \frac{p_o}{\rho_o} \frac{T}{T_o}} \quad (19)$$

true airspeed can be determined from measurements of M (which can be measured directly or computed from measurements of V_c and H) and of free-air temperature T . From the measurements of T , the speed of

sound can be calculated by using any of the following formulas derived from equation (19):

$$a = 33.42353\sqrt{T} \quad (20)$$

where a is in miles per hour and T is in $^{\circ}\text{F}$ absolute

$$a = 29.04425\sqrt{T} \quad (21)$$

where a is in knots and T is in $^{\circ}\text{F}$ absolute

$$a = 44.84237\sqrt{T} \quad (22)$$

where a is in miles per hour and T is in $^{\circ}\text{C}$ absolute

$$a = 38.96695\sqrt{T} \quad (23)$$

where a is in knots and T is in $^{\circ}\text{C}$ absolute

By the use of formulas (20) and (21) values of the speed of sound, in units of miles per hour and knots, have been calculated for a temperature range from -70°F to 120°F and are given in table VIII. With formulas (22) and (23), values of the speed of sound, in units of miles per hour and knots, have been calculated for a temperature range from -60°C to 50°C and are given in table IX.

In table X values of the speed of sound (in both miles per hour and knots) and of free-stream temperature (in both $^{\circ}\text{F}$ and $^{\circ}\text{C}$) in the standard atmosphere are given for altitudes from 0 to 100,000 geopotential feet. These values of the speed of sound and of temperature have been converted from those given in other units in reference 10. As an indication of the manner in which true airspeed V varies with calibrated airspeed V_c and altitude H , values of V have been calculated for V_c from 0 to 1,000 knots and for H from 0 to 100,000 geopotential feet and are given in table XI. These values of V were computed by using the values of q_c , p , q_c/p , and a given in tables III, V, VII, and X, respectively.

It should be noted that the value of T required for the calculation of V is the temperature of the free stream. For the usual case, the temperature measured by an aircraft thermometer is greater than the stream value because of the adiabatic heating effect of the air flow on the

temperature sensor. If the sensor measures the full adiabatic temperature increase (that is, if the recovery factor of the sensor is 1.0) or if the sensor is located in a region where the local velocity of the air is equal to the free-stream velocity, the free-air temperature can be calculated from the following expression:

$$T = \frac{T_m}{1 + \frac{\gamma - 1}{2} \eta M^2} \quad (24)$$

where T_m is the measured temperature and η is the recovery factor of the temperature sensor. For the more general case in which the recovery factor is less than 1.0 and the sensor is located in a region where the local velocity differs from the free-stream value, the free-stream temperature can be computed from the following equation:

$$T = \frac{T_m}{1 + \frac{\gamma - 1}{2} \eta M_1^2} \frac{1 + \frac{\gamma - 1}{2} M_1^2}{1 + \frac{\gamma - 1}{2} M^2} \quad (25)$$

where M_1 is the local stream Mach number, which can be determined from measurements of the local static pressure in the region in which the temperature sensor is located.

In the calibration of true-airspeed instruments utilizing measurements of total pressure, static pressure, and air temperature, the recovery factor of the thermometer installation must, of course, be taken into account. For the values of true airspeed given in table XI, free-stream temperatures (that is, $\eta = 0$) were assumed.

Langley Research Center,
National Aeronautics and Space Administration,
Langley Field, Va., March 8, 1961.

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TABLE I.- IMPACT PRESSURE q_c IN INCHES OF MERCURY FOR VALUES OF CALIBRATED AIRSPEED V_c IN MILES PER HOUR

Calibrated airspeed, V_c , mph	0	1	2	3	4	5	6	7	8	9
0	0.00000	0.00000	0.00014	0.00032	0.00054	0.00081	0.00129	0.00176	0.00231	0.00292
10	.003611	.004377	.005203	.006104	.007088	.008136	.009258	.010451	.011708	.013049
20	.014461	.015939	.017501	.019129	.020825	.022594	.024440	.026358	.028347	.030412
30	.032539	.034751	.037025	.039382	.041806	.044394	.046869	.049508	.052228	.055019
40	.057871	.060806	.063816	.066886	.070004	.073276	.076551	.079920	.083358	.086876
50	.090464	.094126	.097851	.101652	.105535	.109485	.113509	.117603	.121771	.126016
60	.130331	.134712	.139179	.143709	.148323	.152999	.157751	.162574	.167472	.172448
70	.177496	.182606	.187804	.193064	.198411	.203818	.209302	.214859	.220487	.226196
80	.231977	.237832	.243751	.249736	.255824	.261976	.268196	.274489	.280853	.287298
90	.293812	.300397	.307058	.313802	.320606	.327488	.334454	.341485	.348592	.355770
100	.363029	.370347	.377756	.385239	.392785	.400406	.408111	.415888	.423736	.431659
110	.439657	.447733	.455874	.464097	.472391	.480772	.489233	.497771	.506388	.515080
120	.523742	.532566	.541464	.550443	.559480	.568606	.577797	.587070	.596414	.605837
130	.615334	.624912	.634564	.644292	.654095	.663965	.673914	.683940	.694039	.704227
140	.714481	.724804	.735210	.745698	.756263	.766894	.777600	.788389	.799257	.810199
150	.821216	.832111	.843083	.854134	.865260	.876456	.887831	.900480	.912117	.923825
160	.935611	.947472	.959413	.971424	.983524	.995699	1.007951	1.020288	1.032688	1.045156
170	1.05772	1.070307	1.083007	1.095866	1.108773	1.121688	1.134707	1.14782	1.161000	1.17426
180	1.18760	1.20103	1.21451	1.22809	1.24175	1.25548	1.26929	1.28319	1.29716	1.31120
190	1.32533	1.33953	1.35382	1.36819	1.38263	1.39716	1.41176	1.42645	1.44121	1.45605
200	1.47097	1.48597	1.50106	1.51622	1.53146	1.54679	1.56219	1.57768	1.59324	1.60889
210	1.62461	1.64042	1.65630	1.67228	1.68833	1.70445	1.72066	1.73695	1.75333	1.76978
220	1.78631	1.80294	1.81964	1.83642	1.85328	1.87022	1.88725	1.90435	1.92155	1.93882
230	1.95617	1.97362	1.99114	2.00874	2.02643	2.04419	2.06204	2.07998	2.09800	2.11609
240	2.13429	2.15255	2.17090	2.18935	2.20785	2.22645	2.24514	2.26390	2.28276	2.30170
250	2.32071	2.33983	2.35902	2.37829	2.39765	2.41710	2.43662	2.45624	2.47592	2.49572
260	2.51558	2.53555	2.55558	2.57571	2.59592	2.61622	2.63659	2.65706	2.67762	2.69826
270	2.71899	2.73980	2.76070	2.78168	2.80276	2.82392	2.84516	2.86650	2.88792	2.90943
280	2.93102	2.95271	2.97448	2.99634	3.01828	3.04032	3.06244	3.08464	3.10695	3.12935
290	3.15181	3.17436	3.19703	3.21976	3.24260	3.26552	3.28853	3.31163	3.33481	3.35809
300	3.38145	3.40491	3.42845	3.45209	3.47582	3.49963	3.52354	3.54754	3.57163	3.59581
310	3.62008	3.64444	3.66890	3.69343	3.71807	3.74279	3.76761	3.79252	3.81752	3.84262
320	3.86781	3.89308	3.91845	3.94392	3.96947	3.99512	4.02087	4.04670	4.07265	4.09868
330	4.12477	4.15097	4.17728	4.20367	4.23016	4.25674	4.28343	4.31020	4.33707	4.36403
340	4.39109	4.41824	4.44549	4.47282	4.50027	4.52780	4.55544	4.58316	4.61098	4.63889
350	4.66691	4.69502	4.72323	4.75153	4.77993	4.80843	4.83703	4.86572	4.89452	4.92339
360	4.95238	4.98147	5.01064	5.03992	5.06931	5.09878	5.12835	5.15803	5.18780	5.21768
370	5.24764	5.27772	5.30788	5.33816	5.36853	5.39899	5.42957	5.46024	5.49100	5.52187
380	5.55285	5.58392	5.61510	5.64638	5.67773	5.70923	5.74080	5.77249	5.80428	5.83617
390	5.86816	5.90025	5.93244	5.96475	5.99715	6.02965	6.06227	6.09498	6.12780	6.16071
400	6.19373	6.22686	6.26010	6.29343	6.32688	6.36043	6.39407	6.42783	6.46169	6.49566
410	6.52974	6.56393	6.59820	6.63260	6.66710	6.70171	6.73642	6.77124	6.80617	6.84121
420	6.87635	6.91161	6.94696	6.98243	7.01800	7.05367	7.08944	7.12532	7.16130	7.19732
430	7.23373	7.27009	7.30655	7.34310	7.37978	7.41656	7.45344	7.49042	7.52757	7.56479
440	7.60213	7.63958	7.67713	7.71480	7.75258	7.79048	7.82849	7.86660	7.90485	7.94319
450	7.98166	8.02022	8.05891	8.09772	8.13663	8.17566	8.21481	8.25407	8.29345	8.33294
460	8.37254	8.41226	8.45209	8.49205	8.53212	8.57229	8.61261	8.65307	8.69356	8.73424
470	8.77498	8.81587	8.85688	8.89800	8.93924	8.98060	9.02207	9.06368	9.10540	9.14723
480	9.18919	9.23127	9.27346	9.31578	9.35821	9.40077	9.44345	9.48625	9.52917	9.57221
490	9.61537	9.65866	9.70207	9.74560	9.78926	9.83303	9.87693	9.92095	9.96509	10.0094
500	10.0538	10.0983	10.1429	10.1877	10.2326	10.2776	10.3227	10.3680	10.4134	10.4589
510	10.5046	10.5503	10.5962	10.6423	10.6884	10.7347	10.7811	10.8276	10.8743	10.9211
520	10.9680	11.0151	11.0623	11.1096	11.1570	11.2046	11.2523	11.3001	11.3481	11.3962
530	11.4444	11.4927	11.5412	11.5898	11.6386	11.6875	11.7365	11.7856	11.8349	11.8843
540	11.9339	11.9836	12.0334	12.0833	12.1334	12.1836	12.2340	12.2845	12.3351	12.3859
550	12.4367	12.4878	12.5390	12.5903	12.6417	12.6933	12.7450	12.7969	12.8489	12.9010
560	12.9533	13.0057	13.0582	13.1109	13.1638	13.2167	13.2698	13.3231	13.3765	13.4300
570	13.4837	13.5373	13.5915	13.6456	13.6999	13.7542	13.8088	13.8635	13.9183	13.9732
580	14.0283	14.0836	14.1390	14.1945	14.2502	14.3061	14.3620	14.4182	14.4744	14.5309
590	14.5874	14.6441	14.7010	14.7580	14.8152	14.8725	14.9299	14.9875	15.0453	15.1032

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TABLE I.- IMPACT PRESSURE q_c IN INCHES OF MERCURY FOR VALUES OF CALIBRATED AIRSPEED V_c IN MILES PER HOUR - Concluded

Calibrated airspeed, V_c , mph	0	1	2	3	4	5	6	7	8	9
600	15.1613	15.2199	15.2778	15.3363	15.3950	15.4538	15.5128	15.5719	15.6312	15.6906
610	15.7502	15.8099	15.8698	15.9298	15.9900	16.0503	16.1108	16.1715	16.2323	16.2933
620	16.3544	16.4156	16.4771	16.5387	16.6004	16.6623	16.7244	16.7866	16.8490	16.9116
630	16.9743	17.0371	17.1001	17.1633	17.2267	17.2902	17.3538	17.4176	17.4816	17.5458
640	17.6101	17.6746	17.7392	17.8040	17.8690	17.9341	17.9994	18.0649	18.1305	18.1963
650	18.2622	18.3284	18.3946	18.4611	18.5277	18.5945	18.6615	18.7286	18.7959	18.8634
660	18.9310	18.9988	19.0668	19.1349	19.2032	19.2717	19.3404	19.4092	19.4782	19.5474
670	19.6167	19.6863	19.7560	19.8259	19.8959	19.9661	20.0365	20.1070	20.1778	20.2487
680	20.3198	20.3911	20.4625	20.5341	20.6059	20.6779	20.7501	20.8224	20.8949	20.9676
690	21.0409	21.1136	21.1868	21.2602	21.3338	21.4076	21.4816	21.5557	21.6300	21.7046
700	21.7793	21.8541	21.9292	22.0045	22.0799	22.1555	22.2313	22.3073	22.3835	22.4599
710	22.5364	22.6132	22.6901	22.7672	22.8446	22.9222	22.9997	23.0776	23.1557	23.2339
720	23.3124	23.3911	23.4699	23.5489	23.6281	23.7076	23.7872	23.8670	23.9470	24.0272
730	24.1076	24.1881	24.2689	24.3499	24.4311	24.5125	24.5940	24.6758	24.7578	24.8399
740	24.9223	25.0049	25.0877	25.1706	25.2538	25.3372	25.4207	25.5045	25.5885	25.6727
750	25.7571	25.8417	25.9265	26.0115	26.0967	26.1821	26.2677	26.3535	26.4396	26.5258
760	26.6122	26.6989	26.7861	26.8731	26.9604	27.0479	27.1356	27.2235	27.3116	27.4000
770	27.4889	27.5772	27.6661	27.7553	27.8446	27.9341	28.0239	28.1138	28.2040	28.2943
780	28.3848	28.4756	28.5665	28.6577	28.7490	28.8405	28.9323	29.0242	29.1163	29.2086
790	29.3011	29.3939	29.4868	29.5799	29.6732	29.7667	29.8603	29.9542	30.0483	30.1427
800	30.2570	30.3516	30.4464	30.5414	30.6366	30.7320	30.8276	30.9234	30.9994	31.0955
810	31.1918	31.2884	31.3851	31.4820	31.5791	31.6763	31.7738	31.8714	31.9692	32.0672
820	32.1654	32.2638	32.3624	32.4611	32.5600	32.6591	32.7584	32.8579	32.9575	33.0574
830	33.1574	33.2576	33.3579	33.4583	33.5592	33.6601	33.7612	33.8625	33.9639	34.0655
840	34.1674	34.2693	34.3715	34.4738	34.5763	34.6790	34.7819	34.8849	34.9881	35.0915
850	35.1931	35.2968	35.4007	35.5058	35.6111	35.7165	35.8221	35.9279	36.0339	36.1399
860	36.2409	36.3478	36.4548	36.5622	36.6698	36.7776	36.8857	36.9940	37.1025	37.2112
870	37.3027	37.4099	37.5173	37.6248	37.7325	37.8404	37.9484	38.0566	38.1649	38.2733
880	38.3822	38.4910	38.6001	38.7093	38.8187	38.9282	39.0379	39.1478	39.2578	39.3680
890	39.4784	39.5889	39.6996	39.8105	39.9215	40.0327	40.1441	40.2556	40.3673	40.4792
900	40.5912	40.7034	40.8157	40.9283	41.0409	41.1538	41.2668	41.3799	41.4933	41.6068
910	41.7204	41.8342	41.9482	42.0624	42.1767	42.2911	42.4057	42.5205	42.6355	42.7506
920	42.8659	42.9813	43.0969	43.2126	43.3286	43.4446	43.5609	43.6772	43.7938	43.9105
930	44.0274	44.1444	44.2616	44.3790	44.4965	44.6141	44.7320	44.8499	44.9681	45.0864
940	45.2049	45.3233	45.4422	45.5612	45.6803	45.7995	45.9189	46.0385	46.1581	46.2781
950	46.3981	46.5183	46.6386	46.7591	46.8798	47.0006	47.1216	47.2427	47.3640	47.4854
960	47.6078	47.7288	47.8499	47.9727	48.0949	48.2173	48.3398	48.4625	48.5853	48.7083
970	48.8313	48.9547	49.0782	49.2018	49.3256	49.4495	49.5735	49.6977	49.8221	49.9467
980	50.0713	50.1962	50.3211	50.4463	50.5716	50.6970	50.8226	50.9484	51.0743	51.2003
990	51.3263	51.4529	51.5794	51.7061	51.8329	51.9598	52.0870	52.2142	52.3417	52.4692
1,000	52.5970	52.7248	52.8529	52.9810	53.1094	53.2379	53.3665	53.4953	53.6242	53.7533
1,100	55.8825	56.0119	56.1414	56.2711	56.4010	56.5309	56.6611	56.7914	56.9218	57.0524
1,200	59.1831	59.3140	59.4450	59.5762	59.7076	59.8390	59.9707	60.1024	60.2344	60.3665
1,300	62.4987	62.6311	62.7636	62.8963	63.0291	63.1621	63.2952	63.4285	63.5619	63.6954
1,400	65.8292	65.9630	66.0970	66.2312	66.3655	66.4999	66.6345	66.7693	66.9042	67.0392
1,500	69.1744	69.3098	69.4452	69.5809	69.7167	69.8526	69.9887	70.1249	70.2613	70.3978
1,600	72.5344	72.6712	72.8082	72.9453	73.0826	73.2200	73.3575	73.4952	73.6330	73.7710
1,700	75.9091	76.0474	76.1858	76.3244	76.4631	76.6020	76.7410	76.8801	77.0194	77.1589
1,800	79.2983	79.4382	79.5781	79.7181	79.8583	79.9986	80.1391	80.2797	80.4204	80.5613
1,900	82.7023	82.8433	82.9843	83.1253	83.2663	83.4073	83.5483	83.6893	83.8303	83.9713
1,100	66.1208	-----	-----	-----	-----	-----	-----	-----	-----	-----

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TABLE II. - IMPACT PRESSURE q_c IN POUNDS PER SQUARE FOOT FOR VALUES OF CALIBRATED AIRSPEED V_c IN KILOMETERS PER HOUR

Calibrated airspeed, V_c , mph	0	1	2	3	4	5	6	7	8	9
0	0.000000	0.002116	0.010369	0.022895	0.040631	0.063698	0.091844	0.125068	0.163584	0.207389
10	.25427	.30603	.368010	.431708	.501352	.577999	.654758	.739195	.828076	.922882
20	1.02277	1.12731	1.25778	1.39290	1.52899	1.67796	1.78953	1.86418	2.00490	2.15092
30	2.30139	2.45777	2.61861	2.78936	2.92678	3.13348	3.31484	3.50149	3.69386	3.89130
40	4.09298	4.30058	4.51347	4.73099	4.95322	5.18113	5.41415	5.65242	5.89597	6.14444
50	6.39817	6.69720	6.99066	7.18942	7.46411	7.74345	8.02808	8.31758	8.61237	8.91266
60	9.21782	9.52763	9.84358	10.1640	10.4903	10.8211	11.1571	11.4985	11.8447	12.1966
70	12.5536	12.9151	13.2826	13.6547	14.0328	14.4152	14.8022	15.1941	15.5902	15.9980
80	16.4068	16.8210	17.2396	17.6645	18.0934	18.5289	18.9685	19.4135	19.8637	20.3195
90	20.7802	21.2460	21.7170	22.1940	22.6753	23.1620	23.6547	24.1520	24.6546	25.1622
100	25.6736	26.1933	26.7172	27.2465	27.7802	28.3192	28.8641	29.4141	29.9692	30.5296
110	31.0953	31.6664	32.2423	32.8238	33.4104	34.0032	34.6001	35.2028	35.8106	36.4245
120	37.0423	37.6663	38.2957	38.9308	39.5699	40.2133	40.8654	41.5212	42.1821	42.8485
130	43.5202	44.1976	44.8805	45.5689	46.2609	46.9597	47.6653	48.3782	49.0967	49.8207
140	50.5329	51.2626	51.9986	52.7404	53.4879	54.2392	54.9967	55.7602	56.5284	57.3023
150	58.0815	58.8662	59.6564	60.4521	61.2542	62.0729	62.8977	63.7282	64.5650	65.4086
160	66.1722	67.0111	67.8557	68.7051	69.5609	70.4222	71.2882	72.1582	73.0327	73.9112
170	74.8083	75.7024	76.6016	77.5062	78.4162	79.3321	80.2531	81.1806	82.1139	83.0527
180	83.9943	84.9441	85.8979	86.8562	87.8201	88.7895	89.7645	90.7450	91.7312	92.7232
190	93.7335	94.7401	95.7508	96.7666	97.7885	98.8161	99.8486	100.887	101.931	102.981
200	104.036	105.097	106.164	107.236	108.315	109.399	110.488	111.583	112.683	113.789
210	114.903	116.021	117.144	118.274	119.409	120.549	121.696	122.848	124.006	125.170
220	126.339	127.513	128.696	129.885	131.075	132.274	133.478	134.688	135.904	137.125
230	138.353	139.587	140.826	142.071	143.322	144.578	145.841	147.109	148.383	149.663
240	150.950	152.242	153.539	154.843	156.153	157.469	158.790	160.117	161.451	162.790
250	164.135	165.487	166.844	168.208	169.577	170.952	172.333	173.720	175.114	176.513
260	177.918	179.330	180.747	182.170	183.599	185.035	186.476	187.924	189.378	190.837
270	192.304	193.776	195.254	196.738	198.228	199.723	201.222	202.727	204.231	205.733
280	207.300	208.834	210.374	211.919	213.472	215.030	216.595	218.157	219.722	221.286
290	222.915	224.511	226.114	227.722	229.337	230.958	232.585	234.219	235.858	237.505
300	239.157	240.817	242.481	244.153	245.831	247.516	249.207	250.904	252.608	254.318
310	256.035	257.757	259.487	261.222	262.965	264.714	266.469	268.228	269.992	271.774
320	273.356	275.143	276.937	278.739	280.546	282.360	284.181	286.008	287.841	289.682
330	291.129	292.982	294.843	296.710	298.583	300.463	302.349	304.241	306.139	308.042
340	310.365	312.285	314.212	316.146	318.087	320.034	322.009	324.001	326.001	328.007
350	330.073	332.061	334.056	336.058	338.066	340.082	342.105	344.134	346.171	348.213
360	350.263	352.320	354.384	356.455	358.533	360.618	362.709	364.808	366.914	369.026
370	371.146	373.273	375.406	377.548	379.695	381.850	384.013	386.182	388.358	390.541
380	392.732	394.930	397.134	399.347	401.566	403.792	406.025	408.266	410.514	412.770
390	415.032	417.302	419.579	421.864	424.155	426.454	428.761	431.074	433.396	435.724
400	438.059	440.402	442.753	445.110	447.476	449.849	452.228	454.616	457.011	459.414
410	461.824	464.242	466.666	469.099	471.539	473.987	476.442	478.904	481.375	483.853
420	486.338	488.832	491.332	493.841	496.357	498.881	501.413	503.952	506.499	509.053
430	511.616	514.186	516.764	519.350	521.943	524.545	527.155	529.772	532.396	535.029
440	537.670	540.318	542.974	545.639	548.311	550.992	553.679	556.373	559.074	561.782
450	564.513	567.240	569.976	572.721	575.473	578.233	581.002	583.779	586.564	589.357
460	592.158	594.967	597.784	600.610	603.445	606.286	609.137	611.996	614.862	617.737
470	620.621	623.513	626.413	629.322	632.238	635.164	638.097	641.040	643.990	646.949
480	649.916	652.892	655.877	658.870	661.871	664.881	667.899	670.926	673.962	677.006
490	680.059	683.120	686.191	689.269	692.357	695.453	698.558	701.671	704.793	707.925
500	711.065	714.213	717.369	720.536	723.711	726.895	730.088	733.290	736.500	739.719
510	742.948	746.185	749.431	752.687	755.951	759.225	762.507	765.798	769.098	772.408
520	775.726	779.054	782.391	785.737	789.093	792.457	795.830	799.213	802.605	806.007
530	809.418	812.837	816.267	819.705	823.153	826.610	830.077	833.552	837.038	840.533
540	844.037	847.551	851.075	854.607	858.149	861.701	865.263	868.835	872.414	876.004
550	879.604	883.214	886.835	890.462	894.100	897.748	901.406	905.074	908.751	912.438
560	916.136	919.843	923.560	927.287	931.024	934.770	938.526	942.293	946.070	949.856
570	953.632	957.439	961.273	965.102	968.939	972.786	976.642	980.509	984.386	988.274
580	992.171	996.080	999.998	1003.93	1007.86	1011.81	1015.77	1019.74	1023.72	1027.71
590	1031.71	1035.72	1039.75	1043.78	1047.82	1051.87	1055.94	1060.01	1064.10	1068.19

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TABLE III.- IMPACT PRESSURE q_c IN INCHES OF MERCURY FOR VALUES OF CALIBRATED AIRSPEED V_c IN KNOTS

Calibrated airspeed, V_c , knots	0	1	2	3	4	5	6	7	8	9
0	0.000000	0.000051	0.000189	0.000428	0.000763	0.001194	0.001726	0.002346	0.003067	0.003875
10	.004784	.005790	.006891	.008085	.009373	.010756	.012233	.013803	.015468	.017228
20	.019150	.021118	.023171	.025311	.027537	.029850	.032250	.034737	.037311	.040274
30	.043108	.046031	.049047	.052155	.055357	.058653	.062043	.065527	.069105	.072877
40	.076655	.080548	.084528	.088606	.092777	.097047	.101412	.105870	.110430	.115092
50	.119841	.124691	.129640	.134682	.139822	.145052	.150381	.155809	.161337	.166965
60	.172679	.178492	.184417	.190422	.196526	.202732	.209039	.215447	.221956	.228565
70	.235205	.242000	.248888	.255866	.262945	.270120	.277409	.284803	.292291	.299881
80	.307483	.315247	.323108	.331067	.339119	.347281	.355549	.363927	.372411	.380998
90	.389550	.398282	.407121	.416067	.425109	.434250	.443490	.452833	.462277	.471821
100	.481424	.491160	.500993	.510923	.520953	.531078	.541291	.551597	.561996	.572487
110	.582225	.593949	.604783	.615728	.626783	.637948	.649223	.660608	.672103	.683708
120	.694996	.706731	.718562	.730483	.742491	.754585	.766763	.779025	.791371	.803801
130	.816826	.829531	.842403	.855444	.868654	.881931	.895283	.908710	.922211	.935786
140	.948779	.962531	.976394	.990364	1.004442	1.018529	1.032723	1.047023	1.061428	1.075938
150	1.090977	1.105775	1.120653	1.135612	1.150722	1.165951	1.181282	1.196713	1.212244	1.227874
160	1.24347	1.25929	1.27521	1.29125	1.30738	1.32362	1.33996	1.35641	1.37298	1.38963
170	1.40640	1.42327	1.44025	1.45733	1.47452	1.49181	1.50921	1.52671	1.54432	1.56205
180	1.57987	1.59780	1.61584	1.63398	1.65223	1.67059	1.68906	1.70763	1.72631	1.74510
190	1.76400	1.78300	1.80211	1.82133	1.84066	1.86009	1.87964	1.89930	1.91905	1.93893
200	1.95891	1.97900	1.99920	2.01951	2.03992	2.06045	2.08108	2.10183	2.12269	2.14366
210	2.16473	2.18593	2.20722	2.22864	2.25016	2.27179	2.29354	2.31539	2.33735	2.35944
220	2.38162	2.40392	2.42634	2.44887	2.47151	2.49426	2.51713	2.54011	2.56320	2.58641
230	2.60972	2.63315	2.65670	2.68036	2.70412	2.72802	2.75202	2.77614	2.80037	2.82471
240	2.84918	2.87375	2.89845	2.92325	2.94818	2.97321	2.99838	3.02365	3.04904	3.07455
250	3.10015	3.12590	3.15176	3.17773	3.20381	3.23003	3.25636	3.28281	3.30937	3.33605
260	3.36284	3.38977	3.41680	3.44396	3.47124	3.49864	3.52613	3.55378	3.58154	3.60941
270	3.63741	3.66533	3.69337	3.72152	3.75006	3.77921	3.80792	3.83678	3.86574	3.89483
280	3.92404	3.95337	3.98281	4.01241	4.04212	4.07194	4.10189	4.13197	4.16216	4.19250
290	4.22293	4.25351	4.28421	4.31503	4.34597	4.37704	4.40825	4.43957	4.47102	4.50260
300	4.53430	4.56613	4.59809	4.63017	4.66238	4.69473	4.72719	4.75978	4.79252	4.82537
310	4.85834	4.89146	4.92469	4.95806	4.99156	5.02519	5.05896	5.09284	5.12687	5.16101
320	5.19529	5.22971	5.26425	5.29892	5.33374	5.36868	5.40375	5.43896	5.47430	5.50977
330	5.54538	5.58111	5.61699	5.65300	5.68914	5.72542	5.76183	5.79838	5.83506	5.87187
340	5.90883	5.94592	5.98315	6.02051	6.05801	6.09565	6.13343	6.17135	6.20939	6.24758
350	6.28590	6.32438	6.36298	6.40173	6.44061	6.47963	6.51880	6.55811	6.59755	6.63713
360	6.67687	6.71674	6.75672	6.79690	6.83729	6.87784	6.91852	6.95934	6.99981	7.04082
370	7.08198	7.12338	7.16472	7.20631	7.24804	7.28991	7.33194	7.37412	7.41643	7.45889
380	7.50151	7.54427	7.58717	7.63021	7.67342	7.71677	7.76027	7.80391	7.84770	7.89165
390	7.93573	7.98000	8.02439	8.06894	8.11363	8.15847	8.20347	8.24863	8.29393	8.33939
400	8.38499	8.43073	8.47668	8.52274	8.56897	8.61535	8.66188	8.70856	8.75540	8.80241
410	8.84955	8.89687	8.94434	8.99196	9.03974	9.08768	9.13578	9.18403	9.23244	9.28102
420	9.32975	9.37864	9.42769	9.47691	9.52628	9.57581	9.62551	9.67536	9.72538	9.77556
430	9.82591	9.87640	9.92708	9.97791	10.0289	10.0801	10.1314	10.1829	10.2345	10.2864
440	10.3384	10.3905	10.4428	10.4953	10.5480	10.6008	10.6538	10.7070	10.7603	10.8138
450	10.8675	10.9213	10.9753	11.0295	11.0838	11.1383	11.1930	11.2479	11.3029	11.3582
460	11.4135	11.4691	11.5248	11.5807	11.6368	11.6931	11.7495	11.8061	11.8629	11.9199
470	11.9770	12.0343	12.0918	12.1495	12.2074	12.2654	12.3236	12.3820	12.4406	12.4993
480	12.5583	12.6174	12.6767	12.7362	12.7958	12.8557	12.9157	12.9759	13.0363	13.0969
490	13.1577	13.2186	13.2798	13.3411	13.4026	13.4643	13.5262	13.5883	13.6505	13.7130

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TABLE III. - IMPACT PRESSURE q_c IN INCHES OF MERCURY FOR VALUES OF CALIBRATED AIRSPEED V_c IN KNOTS - Concluded

Calibrated airspeed, V_c , knots	0	1	2	3	4	5	6	7	8	9
500	13.7756	13.8385	13.9015	13.9647	14.0281	14.0917	14.1555	14.2195	14.2836	14.3480
510	14.4126	14.4773	14.5423	14.6074	14.6727	14.7383	14.8040	14.8699	14.9361	15.0024
520	15.0689	15.1356	15.2026	15.2697	15.3370	15.4045	15.4722	15.5402	15.6083	15.6766
530	15.7451	15.8139	15.8828	15.9519	16.0213	16.0908	16.1606	16.2305	16.3007	16.3711
540	16.4417	16.5125	16.5835	16.6547	16.7261	16.7977	16.8695	16.9416	17.0138	17.0863
550	17.1590	17.2319	17.3050	17.3783	17.4518	17.5256	17.5996	17.6737	17.7481	17.8228
560	17.8976	17.9726	18.0479	18.1234	18.1991	18.2750	18.3512	18.4275	18.5041	18.5809
570	18.6580	18.7352	18.8127	18.8904	18.9683	19.0465	19.1248	19.2034	19.2823	19.3613
580	19.4406	19.5201	19.5999	19.6798	19.7600	19.8405	19.9211	20.0020	20.0831	20.1645
590	20.2461	20.3279	20.4099	20.4922	20.5748	20.6575	20.7405	20.8238	20.9072	20.9909
600	21.0749	21.1591	21.2435	21.3282	21.4131	21.4982	21.5836	21.6693	21.7551	21.8413
610	21.9276	22.0142	22.1011	22.1882	22.2755	22.3631	22.4510	22.5391	22.6274	22.7160
620	22.8048	22.8939	22.9833	23.0729	23.1627	23.2528	23.3432	23.4338	23.5246	23.6158
630	23.7071	23.7987	23.8906	23.9828	24.0752	24.1679	24.2608	24.3540	24.4474	24.5411
640	24.6351	24.7293	24.8238	24.9186	25.0136	25.1089	25.2044	25.3003	25.3964	25.4927
650	25.5893	25.6862	25.7834	25.8809	25.9786	26.0765	26.1748	26.2733	26.3721	26.4712
660	26.5705	26.6702	26.7701	26.8703	26.9707	27.0714	27.1724	27.2737	27.3753	27.4771
670	27.5792	27.6813	27.7842	27.8871	27.9902	28.0937	28.1974	28.3013	28.4056	28.5101
680	28.6148	28.7198	28.8251	28.9306	29.0364	29.1425	29.2488	29.3554	29.4622	29.5693
690	29.6767	29.7843	29.8922	30.0003	30.1086	30.2173	30.3261	30.4353	30.5447	30.6543
700	30.7642	30.8743	30.9847	31.0953	31.2062	31.3173	31.4287	31.5403	31.6522	31.7643
710	31.8766	31.9892	32.1021	32.2151	32.3285	32.4421	32.5559	32.6699	32.7842	32.8988
720	33.0133	33.1285	33.2438	33.3593	33.4750	33.5910	33.7072	33.8236	33.9403	34.0572
730	34.1744	34.2918	34.4094	34.5272	34.6453	34.7636	34.8822	35.0010	35.1200	35.2393
740	35.3587	35.4785	35.5984	35.7186	35.8390	35.9596	36.0805	36.2016	36.3229	36.4444
750	36.5662	36.6882	36.8104	36.9329	37.0555	37.1785	37.3016	37.4249	37.5485	37.6723
760	37.7954	37.9206	38.0451	38.1698	38.2947	38.4198	38.5452	38.6708	38.7966	38.9226
770	39.0489	39.1754	39.3021	39.4290	39.5561	39.6835	39.8110	39.9388	40.0668	40.1951
780	40.3235	40.4522	40.5811	40.7102	40.8395	40.9690	41.0988	41.2287	41.3589	41.4893
790	41.6199	41.7507	41.8818	42.0130	42.1445	42.2762	42.4081	42.5402	42.6725	42.8051
800	42.9378	43.0708	43.2040	43.3374	43.4710	43.6048	43.7388	43.8730	44.0073	44.1422
810	44.2770	44.4121	44.5474	44.6829	44.8186	44.9545	45.0907	45.2270	45.3635	45.5003
820	45.6373	45.7744	45.9118	46.0494	46.1872	46.3252	46.4634	46.6019	46.7405	46.8793
830	47.0184	47.1576	47.2971	47.4367	47.5766	47.7167	47.8569	47.9974	48.1381	48.2790
840	48.4201	48.5614	48.7029	48.8446	48.9866	49.1287	49.2710	49.4135	49.5563	49.6992
850	49.8423	49.9857	50.1292	50.2730	50.4169	50.5611	50.7055	50.8500	50.9948	51.1397
860	51.2849	51.4303	51.5758	51.7216	51.8676	52.0138	52.1601	52.3067	52.4535	52.6004
870	52.7476	52.8950	53.0426	53.1904	53.3383	53.4865	53.6349	53.7834	53.9322	54.0812
880	54.2304	54.3798	54.5293	54.6791	54.8291	54.9792	55.1296	55.2801	55.4309	55.5819
890	55.7330	55.8844	56.0359	56.1877	56.3396	56.4918	56.6441	56.7966	56.9494	57.1023
900	57.2534	57.4088	57.5623	57.7160	57.8699	58.0240	58.1783	58.3328	58.4875	58.6424
910	58.7973	58.9528	59.1083	59.2640	59.4198	59.5759	59.7321	59.8886	60.0453	60.2021
920	60.3591	60.5164	60.6738	60.8314	60.9892	61.1473	61.3055	61.4639	61.6225	61.7812
930	61.9408	62.0994	62.2588	62.4183	62.5781	62.7380	62.8982	63.0585	63.2190	63.3798
940	63.5407	63.7018	63.8631	64.0246	64.1862	64.3481	64.5102	64.6725	64.8349	64.9976
950	65.1604	65.3234	65.4867	65.6500	65.8137	65.9774	66.1414	66.3056	66.4700	66.6346
960	66.7993	66.9643	67.1294	67.2947	67.4602	67.6259	67.7918	67.9579	68.1242	68.2907
970	68.4573	68.6242	68.7912	68.9585	69.1259	69.2935	69.4613	69.6293	69.7975	69.9659
980	70.1344	70.3032	70.4721	70.6413	70.8106	70.9801	71.1498	71.3197	71.4897	71.6600
990	71.8305	72.0011	72.1719	72.3430	72.5142	72.6856	72.8572	73.0290	73.2009	73.3731
1,000	73.5454	-----	-----	-----	-----	-----	-----	-----	-----	-----

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TABLE IV.-- IMPACT PRESSURE q_c IN POUNDS PER SQUARE FOOT FOR VALUES OF CALIBRATED AIRSPEED V_c IN KNOTS

Calibrated airspeed, V_c , knots	0	1	2	3	4	5	6	7	8	9
0	0.000000	0.003958	0.013332	0.030262	0.053954	0.084437	0.122206	0.167911	0.216912	0.274090
10	1.38385	1.409488	1.87365	1.71802	663646	761415	866591	978327	1.09684	1.22233
20	1.59436	1.49365	1.63880	1.79159	1.95075	2.11685	2.28974	2.46899	2.65386	2.84483
30	3.04885	3.25559	3.46890	3.68941	3.91648	4.14990	4.38915	4.63496	4.88748	5.14582
40	5.42154	5.69686	5.97831	6.26675	6.56374	6.86970	7.17499	7.47981	7.78432	8.08865
50	8.47587	8.81891	9.16893	9.52552	9.88908	10.25900	10.63499	11.01628	11.40215	11.80000
60	12.2129	12.6241	13.0431	13.4678	13.8982	14.3344	14.7765	15.2248	15.6794	16.1404
70	16.6352	17.1158	17.6029	18.0964	18.5964	19.1030	19.6162	20.1361	20.6628	21.2045
80	21.7471	22.2965	22.8522	23.4151	23.9864	24.5661	25.1543	25.7511	26.3566	26.9708
90	27.5900	28.1690	28.7541	29.3468	29.9481	30.5581	31.1768	31.8041	32.4401	33.0849
100	34.0495	34.7379	35.4353	36.1427	36.8600	37.5872	38.3243	39.0714	39.8285	40.5956
110	41.2495	42.0078	42.7740	43.5481	44.3300	45.1207	45.9201	46.7282	47.5450	48.3705
120	49.1544	49.9844	50.8212	51.6649	52.5155	53.3731	54.2376	55.1090	55.9873	56.8724
130	57.7710	58.6717	59.5800	60.4952	61.4175	62.3471	63.2838	64.2276	65.1784	66.1361
140	67.1035	68.0762	69.0566	70.0447	71.0401	72.0431	73.0538	74.0721	75.0980	76.1314
150	77.1598	78.2056	79.2576	80.3167	81.3831	82.4567	83.5376	84.6258	85.7214	86.8244
160	87.9462	89.0650	90.1911	91.3243	92.4648	93.6127	94.7680	95.9316	97.1036	98.2840
170	99.4696	100.662	101.865	103.077	104.297	105.524	106.758	107.998	109.244	110.496
180	111.748	113.006	114.282	115.565	116.856	118.155	119.460	120.774	122.096	123.424
190	124.761	126.105	127.456	128.816	130.183	131.557	132.939	134.329	135.727	137.133
200	138.546	139.967	141.396	142.832	144.276	145.728	147.187	148.653	150.126	151.605
210	153.105	154.602	156.108	157.625	159.145	160.670	162.201	163.737	165.278	166.824
220	168.443	170.020	171.606	173.199	174.800	176.410	178.027	179.652	181.285	182.927
230	184.576	186.235	187.898	189.572	191.252	192.942	194.640	196.346	198.059	199.781
240	201.511	203.250	205.006	206.770	208.545	210.328	212.118	213.915	215.717	217.524
250	219.262	221.083	222.912	224.749	226.594	228.448	230.310	232.181	234.059	235.946
260	237.841	239.746	241.657	243.578	245.507	247.445	249.391	251.345	253.307	255.280
270	257.260	259.249	261.246	263.252	265.266	267.289	269.320	271.361	273.409	275.467
280	277.535	279.607	281.691	283.782	285.884	287.993	290.111	292.238	294.374	296.519
290	298.672	300.835	303.008	305.186	307.374	309.572	311.779	313.994	316.218	318.452
300	320.694	322.945	325.205	327.474	329.753	332.040	334.336	336.641	338.956	341.280
310	343.612	345.954	348.304	350.665	353.034	355.415	357.801	360.197	362.604	365.019
320	367.445	369.877	372.320	374.775	377.239	379.706	382.187	384.677	387.177	389.685
330	392.205	394.731	397.269	399.815	402.371	404.937	407.512	410.097	412.692	415.295
340	417.909	420.535	423.166	425.808	428.460	431.122	433.794	436.476	439.166	441.867
350	444.578	447.300	450.030	452.770	455.520	458.280	461.050	463.830	466.620	469.419
360	472.230	475.049	477.879	480.719	483.569	486.429	489.299	492.179	495.070	497.971
370	500.881	503.802	506.733	509.675	512.626	515.588	518.560	521.545	524.536	527.539
380	530.555	533.577	536.612	539.656	542.712	545.778	548.854	551.941	555.038	558.147
390	561.285	564.355	567.435	570.526	573.626	576.736	579.854	582.981	586.118	589.265
400	593.059	596.275	599.525	602.781	606.050	609.331	612.622	615.923	619.236	622.561
410	627.895	629.241	632.599	635.967	639.346	642.737	646.139	649.553	652.976	656.412
420	659.858	663.316	666.785	670.266	673.757	677.261	680.776	684.301	687.839	691.388
430	694.949	698.500	702.105	705.700	709.307	712.925	716.554	720.197	723.850	727.516
440	731.195	734.881	738.582	742.294	746.018	749.754	753.502	757.262	761.034	764.818
450	768.615	772.421	776.241	780.073	783.917	787.774	791.641	795.522	799.414	803.320
460	807.337	811.166	815.108	819.062	823.029	827.008	831.000	835.005	839.020	843.049
470	847.090	851.145	855.210	859.289	863.381	867.485	871.602	875.732	879.875	884.030
480	888.159	892.375	896.573	900.780	905.000	909.232	913.479	917.737	922.009	926.294
490	930.591	934.905	939.227	943.564	947.915	952.270	956.636	961.014	965.400	969.797

TABLE IV. - IMPACT PRESSURE q_c IN POUNDS PER SQUARE FOOT FOR VALUES OF CALIBRATED AIRSPEED V_c IN KNOTS - Concluded

Calibrated airspeed, V_c , knots	0	1	2	3	4	5	6	7	8	9
500	974.298	978.741	983.199	987.669	992.154	996.651	1001.16	1005.69	1010.23	1014.78
510	1019.35	1023.93	1028.52	1033.13	1037.75	1042.38	1047.03	1051.69	1056.37	1061.06
520	1064.41	1069.03	1073.68	1078.35	1083.04	1087.74	1092.46	1097.19	1101.93	1106.68
530	1109.47	1114.14	1118.83	1123.54	1128.27	1133.02	1137.78	1142.56	1147.35	1152.15
540	1154.53	1159.24	1163.98	1168.74	1173.52	1178.32	1183.13	1187.95	1192.78	1197.62
550	1200.00	1204.81	1209.64	1214.49	1219.36	1224.25	1229.15	1234.07	1239.00	1243.94
560	1245.85	1250.71	1255.59	1260.48	1265.39	1270.31	1275.24	1280.18	1285.13	1290.09
570	1301.61	1306.51	1311.43	1316.36	1321.31	1326.27	1331.24	1336.22	1341.21	1346.21
580	1357.36	1362.31	1367.28	1372.26	1377.25	1382.25	1387.26	1392.28	1397.31	1402.35
590	1413.11	1418.11	1423.13	1428.16	1433.20	1438.25	1443.31	1448.38	1453.46	1458.55
600	1468.86	1473.91	1478.98	1484.06	1489.15	1494.25	1499.36	1504.48	1509.61	1514.75
610	1524.61	1529.71	1534.83	1539.96	1545.10	1550.25	1555.41	1560.58	1565.76	1570.95
620	1580.36	1585.51	1590.68	1595.86	1601.05	1606.25	1611.46	1616.68	1621.91	1627.15
630	1636.11	1641.31	1646.53	1651.76	1657.00	1662.25	1667.51	1672.78	1678.06	1683.35
640	1691.86	1697.11	1702.38	1707.66	1712.95	1718.25	1723.56	1728.88	1734.21	1739.55
650	1747.61	1752.91	1758.23	1763.56	1768.90	1774.25	1779.61	1784.98	1790.36	1795.75
660	1803.36	1808.71	1814.08	1819.46	1824.85	1830.25	1835.66	1841.08	1846.51	1851.95
670	1867.61	1873.01	1878.43	1883.86	1889.30	1894.75	1900.21	1905.68	1911.16	1916.65
680	1931.86	1937.31	1942.78	1948.26	1953.75	1959.25	1964.76	1970.28	1975.81	1981.35
690	1995.61	2001.11	2006.63	2012.16	2017.70	2023.25	2028.81	2034.38	2039.96	2045.55
700	2069.36	2074.91	2080.48	2086.06	2091.65	2097.25	2102.86	2108.48	2114.11	2119.75
710	2143.11	2148.71	2154.33	2159.96	2165.60	2171.25	2176.91	2182.58	2188.26	2193.95
720	2216.86	2222.51	2228.18	2233.86	2239.55	2245.25	2250.96	2256.68	2262.41	2268.15
730	2290.61	2296.31	2302.03	2307.76	2313.50	2319.25	2325.01	2330.78	2336.56	2342.35
740	2364.36	2370.11	2375.88	2381.66	2387.45	2393.25	2399.06	2404.88	2410.71	2416.55
750	2438.11	2443.91	2449.73	2455.56	2461.40	2467.25	2473.11	2478.98	2484.86	2490.75
760	2511.86	2517.71	2523.58	2529.46	2535.35	2541.25	2547.16	2553.08	2559.01	2564.95
770	2585.61	2591.51	2597.43	2603.36	2609.30	2615.25	2621.21	2627.18	2633.16	2639.15
780	2659.36	2665.31	2671.28	2677.26	2683.25	2689.25	2695.26	2701.28	2707.31	2713.35
790	2733.11	2739.11	2745.13	2751.16	2757.20	2763.25	2769.31	2775.38	2781.46	2787.55
800	2806.86	2812.91	2818.98	2825.06	2831.15	2837.25	2843.36	2849.48	2855.61	2861.75
810	2880.61	2886.71	2892.83	2898.96	2905.10	2911.25	2917.41	2923.58	2929.76	2935.95
820	2954.36	2960.51	2966.68	2972.86	2979.05	2985.25	2991.46	2997.68	3003.91	3010.15
830	3028.11	3034.31	3040.53	3046.76	3053.00	3059.25	3065.51	3071.78	3078.06	3084.35
840	3101.86	3108.11	3114.38	3120.66	3126.95	3133.25	3139.56	3145.88	3152.21	3158.55
850	3175.61	3181.91	3188.23	3194.56	3200.90	3207.25	3213.61	3219.98	3226.36	3232.75
860	3249.36	3255.71	3262.08	3268.46	3274.85	3281.25	3287.66	3294.08	3300.51	3306.95
870	3323.11	3329.51	3335.93	3342.36	3348.80	3355.25	3361.71	3368.18	3374.66	3381.15
880	3396.86	3403.31	3409.78	3416.26	3422.75	3429.25	3435.76	3442.28	3448.81	3455.35
890	3470.61	3477.11	3483.63	3490.16	3496.70	3503.25	3509.81	3516.38	3522.96	3529.55
900	3544.36	3550.91	3557.48	3564.06	3570.65	3577.25	3583.86	3590.48	3597.11	3603.75
910	3618.11	3624.71	3631.33	3637.96	3644.60	3651.25	3657.91	3664.58	3671.26	3677.95
920	3691.86	3698.51	3705.18	3711.86	3718.55	3725.25	3731.96	3738.68	3745.41	3752.15
930	3765.61	3772.31	3779.03	3785.76	3792.50	3799.25	3806.01	3812.78	3819.56	3826.35
940	3839.36	3846.11	3852.88	3859.66	3866.45	3873.25	3880.06	3886.88	3893.71	3900.55
950	3913.11	3919.91	3926.73	3933.56	3940.40	3947.25	3954.11	3961.08	3968.06	3975.05
960	3986.86	3993.71	4000.58	4007.46	4014.35	4021.25	4028.16	4035.08	4042.01	4048.95
970	4060.61	4067.51	4074.43	4081.36	4088.30	4095.25	4102.21	4109.18	4116.16	4123.15
980	4094.36	4101.31	4108.28	4115.26	4122.25	4129.25	4136.26	4143.28	4150.31	4157.35
990	4168.11	4175.11	4182.13	4189.16	4196.20	4203.25	4210.31	4217.38	4224.46	4231.55
1,000	4241.86	4248.91	4255.98	4263.06	4270.15	4277.25	4284.36	4291.48	4298.61	4305.75

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TABLE V.- STATIC PRESSURE p IN INCHES OF MERCURY FOR VALUES OF
PRESSURE ALTITUDE H IN GEOPOTENTIAL FEET

[These values are from refs. 8 and 10]

Pressure altitude, H , geopotential ft	0	100	200	300	400	500	600	700	800	900
-1,000	31.0185									
0		30.0295	30.1382	30.2471	30.3563	30.4659	30.5758	30.6860	30.7965	30.9073
1,000	29.9213	29.8133	29.7056	29.5983	29.4913	29.3846	29.2782	29.1721	29.0665	28.9608
2,000	28.8557	28.7508	28.6463	28.5421	28.4382	28.3345	28.2312	28.1282	28.0255	27.9231
3,000	27.8210	27.7193	27.6178	27.5166	27.4157	27.3151	27.2148	27.1148	27.0151	26.9158
4,000	26.8167	26.7179	26.6193	26.5211	26.4232	26.3256	26.2283	26.1312	26.0345	25.9380
5,000	25.8418	25.7459	25.6504	25.5550	25.4600	25.3653	25.2708	25.1767	25.0828	24.9892
6,000	24.8959	24.8029	24.7101	24.6177	24.5255	24.4336	24.3419	24.2506	24.1595	24.0687
7,000	23.9782	23.8880	23.7980	23.7083	23.6189	23.5297	23.4409	23.3523	23.2639	23.1759
8,000	23.0881	23.0006	22.9133	22.8263	22.7396	22.6532	22.5670	22.4811	22.3954	22.3100
9,000	22.2249	22.1401	22.0555	21.9711	21.8871	21.8033	21.7197	21.6364	21.5534	21.4706
10,000	21.3881	21.3058	21.2238	21.1421	21.0606	20.9793	20.8983	20.8176	20.7371	20.6569
11,000	20.5769	20.4972	20.4177	20.3385	20.2595	20.1808	20.1023	20.0241	19.9461	19.8684
12,000	19.7909	19.7136	19.6366	19.5599	19.4834	19.4071	19.3310	19.2553	19.1797	19.1044
13,000	19.0293	18.9545	18.8799	18.8055	18.7314	18.6575	18.5839	18.5105	18.4373	18.3644
14,000	18.2917	18.2192	18.1470	18.0749	18.0032	17.9316	17.8603	17.7892	17.7184	17.6477
15,000	17.5773	17.5072	17.4372	17.3675	17.2980	17.2287	17.1597	17.0909	17.0223	16.9539
16,000	16.8858	16.8178	16.7501	16.6827	16.6154	16.5483	16.4815	16.4149	16.3485	16.2824
17,000	16.2164	16.1507	16.0851	16.0198	15.9547	15.8899	15.8252	15.7608	15.6965	15.6325
18,000	15.5687	15.5051	15.4417	15.3785	15.3155	15.2528	15.1902	15.1279	15.0657	15.0038
19,000	14.9421	14.8806	14.8192	14.7581	14.6972	14.6365	14.5760	14.5157	14.4556	14.3957
20,000	14.3360	14.2765	14.2173	14.1582	14.0993	14.0406	13.9821	13.9238	13.8657	13.8076
21,000	13.7501	13.6926	13.6352	13.5781	13.5212	13.4644	13.4079	13.3516	13.2954	13.2394
22,000	13.1836	13.1281	13.0727	13.0175	12.9624	12.9076	12.8530	12.7985	12.7442	12.6902
23,000	12.6363	12.5826	12.5290	12.4757	12.4225	12.3696	12.3168	12.2642	12.2117	12.1595
24,000	12.1074	12.0556	12.0039	11.9523	11.9010	11.8498	11.7988	11.7480	11.6974	11.6469
25,000	11.5967	11.5466	11.4966	11.4469	11.3973	11.3479	11.2987	11.2496	11.2007	11.1520
26,000	11.1035	11.0551	11.0069	10.9589	10.9110	10.8634	10.8158	10.7685	10.7213	10.6743
27,000	10.6274	10.5808	10.5342	10.4879	10.4417	10.3957	10.3498	10.3041	10.2586	10.2133
28,000	10.1681	10.1230	10.0781	10.0334	9.98887	9.94447	9.90023	9.85616	9.81224	9.76848
29,000	9.72488	9.68144	9.63815	9.59502	9.55205	9.50923	9.46658	9.42407	9.38172	9.33952
30,000	9.29748	9.25559	9.21385	9.17227	9.13083	9.08956	9.04843	9.00745	8.96662	8.92594
31,000	8.88541	8.84503	8.80480	8.76472	8.72479	8.68500	8.64536	8.60587	8.56652	8.52732
32,000	8.48826	8.44935	8.41059	8.37197	8.33349	8.29515	8.25696	8.21891	8.18100	8.14324
33,000	8.10561	8.06813	8.03079	7.99358	7.95652	7.91960	7.88281	7.84616	7.80966	7.77328
34,000	7.73705	7.70095	7.66499	7.62917	7.59348	7.55793	7.52251	7.48722	7.45208	7.41706
35,000	7.38218	7.34743	7.31281	7.27833	7.24397	7.20975	7.17566	7.14170	7.10787	7.07417
36,000	7.04060	7.00716	6.97385	6.94066	6.90761	6.87468	6.84188	6.80920	6.77665	6.74423
37,000	6.71194	6.67977	6.64774	6.61584	6.58414	6.55264	6.52134	6.49024	6.45934	6.42864
38,000	6.39698	6.36598	6.33518	6.30448	6.27388	6.24338	6.21298	6.18268	6.15248	6.12238
39,000	6.09679	6.06679	6.03689	6.00709	5.97739	5.94769	5.91809	5.88859	5.85919	5.82989
40,000	5.81069	5.78169	5.75269	5.72369	5.69469	5.66569	5.63669	5.60769	5.57869	5.54969
41,000	5.53801	5.50901	5.48001	5.45101	5.42201	5.39301	5.36401	5.33501	5.30601	5.27701
42,000	5.27813	5.24913	5.22013	5.19113	5.16213	5.13313	5.10413	5.07513	5.04613	5.01713
43,000	5.03045	5.00145	4.97245	4.94345	4.91445	4.88545	4.85645	4.82745	4.79845	4.76945
44,000	4.79439	4.76539	4.73639	4.70739	4.67839	4.64939	4.62039	4.59139	4.56239	4.53339
45,000	4.56940	4.54040	4.51140	4.48240	4.45340	4.42440	4.39540	4.36640	4.33740	4.30840
46,000	4.35497	4.32597	4.29697	4.26797	4.23897	4.20997	4.18097	4.15197	4.12297	4.09397
47,000	4.15061	4.12161	4.09261	4.06361	4.03461	4.00561	3.97661	3.94761	3.91861	3.88961
48,000	3.95584	3.92684	3.89784	3.86884	3.83984	3.81084	3.78184	3.75284	3.72384	3.69484
49,000	3.77020	3.74120	3.71220	3.68320	3.65420	3.62520	3.59620	3.56720	3.53820	3.50920
50,000	3.59328	3.56428	3.53528	3.50628	3.47728	3.44828	3.41928	3.39028	3.36128	3.33228

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TABLE V.- STATIC PRESSURE p IN INCHES OF MERCURY FOR VALUES OF
 PRESSURE ALTITUDE H IN GEOPOTENTIAL FEET - Concluded
 [These values are from refs. 8 and 10]

Pressure altitude, H , geopotential ft	0	200	400	500	600	800
50,000	3.42466	3.39189	3.35945		3.32731	3.29548
51,000	3.26395	3.23273	3.20180		3.17117	3.14083
52,000	3.11078	3.08103	3.05155		3.02236	2.99344
53,000	2.96481	2.93644	2.90835		2.88053	2.85297
54,000	2.82568	2.79864	2.77187		2.74535	2.71909
55,000	2.69308	2.66731	2.64180		2.61652	2.59149
56,000	2.56670	2.54215	2.51783		2.49374	2.46988
57,000	2.44625	2.42285	2.39967		2.37672	2.35398
58,000	2.33146	2.30916	2.28706		2.26519	2.24351
59,000	2.22205	2.20079	2.17974		2.15889	2.13823
60,000	2.11778	2.09752	2.07745		2.05758	2.03789
61,000	2.01840	1.99909	1.97996		1.96102	1.94226
62,000	1.92368	1.90528	1.88705		1.86900	1.85112
63,000	1.83341	1.81587	1.79850		1.78129	1.76425
64,000	1.74737	1.73066	1.71410		1.69770	1.68146
65,000	1.66538	1.64944	1.63366		1.61804	1.60256
66,000	1.5872			1.5495		
67,000	1.5127			1.4768		
68,000	1.4417			1.4075		
69,000	1.3741			1.3415		
70,000	1.3096			1.2785		
71,000	1.2482			1.2185		
72,000	1.1896			1.1613		
73,000	1.1338			1.1068		
74,000	1.0806			1.0549		
75,000	1.0298			1.0054		
76,000	.98152			.95822		
77,000	.93546			.91325		
78,000	.89156			.87039		
79,000	.84973			.82955		
80,000	.80985			.79062		
81,000	.77185			.75352		
82,000	.73563			.71818		
83,000	.70117			.68461		
84,000	.66847			.65274		
85,000	.63742			.62248		
86,000	.60793			.59374		
87,000	.57992			.56644		
88,000	.55330			.54050		
89,000	.52801			.51584		
90,000	.50398			.49241		
91,000	.48113			.47012		
92,000	.45940			.44893		
93,000	.43873			.42878		
94,000	.41907			.40960		
95,000	.40037			.39136		
96,000	.38257			.37399		
97,000	.36563			.35746		
98,000	.34950			.34172		
99,000	.33414			.32674		
100,000	.31951					

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TABLE VI. - STATIC PRESSURE p IN POUNDS PER SQUARE FOOT FOR VALUES OF PRESSURE ALTITUDE H IN GEOPOTENTIAL FEET

[These values are from refs. 8 and 10]

Pressure altitude, H , geopotential ft	0	100	200	300	400	500	600	700	800	900
-1,000	2193.82	2123.87	2131.56	2139.26	2146.99	2154.73	2162.51	2170.30	2178.12	2185.56
0	2116.22	2108.58	2100.97	2093.37	2085.80	2078.26	2070.73	2063.23	2055.75	2048.29
1,000	2040.85	2033.44	2026.04	2018.67	2011.32	2003.99	1996.69	1989.40	1982.14	1974.90
2,000	1967.68	1960.48	1953.30	1946.14	1939.01	1931.89	1924.80	1917.73	1910.68	1903.65
3,000	1896.64	1889.65	1882.68	1875.74	1868.81	1861.91	1855.02	1848.16	1841.32	1834.50
4,000	1827.69	1820.91	1814.15	1807.41	1800.69	1793.99	1787.31	1780.65	1774.01	1767.39
5,000	1760.79	1754.21	1747.65	1741.11	1734.59	1728.09	1721.61	1715.15	1708.71	1702.29
6,000	1695.89	1689.50	1683.14	1676.80	1670.47	1664.17	1657.88	1651.62	1645.37	1639.14
7,000	1632.93	1626.74	1620.57	1614.42	1608.29	1602.17	1596.08	1590.00	1583.94	1577.90
8,000	1571.88	1565.88	1559.90	1553.93	1547.99	1542.06	1536.15	1530.26	1524.39	1518.53
9,000	1512.70	1506.88	1501.08	1495.30	1489.53	1483.79	1478.06	1472.35	1466.66	1460.98
10,000	1455.33	1449.69	1444.07	1438.47	1432.88	1427.31	1421.76	1416.23	1410.71	1405.21
11,000	1399.73	1394.27	1388.82	1383.39	1377.98	1372.59	1367.21	1361.85	1356.51	1351.18
12,000	1345.87	1340.58	1335.30	1330.04	1324.80	1319.58	1314.37	1309.18	1304.00	1298.84
13,000	1293.70	1288.57	1283.46	1278.37	1273.30	1268.23	1263.19	1258.16	1253.15	1248.16
14,000	1243.18	1238.21	1233.27	1228.34	1223.42	1218.52	1213.64	1208.77	1203.92	1199.09
15,000	1194.27	1189.46	1184.67	1179.90	1175.14	1170.40	1165.67	1160.96	1156.27	1151.59
16,000	1146.92	1142.27	1137.64	1133.02	1128.42	1123.83	1119.26	1114.70	1110.15	1105.63
17,000	1101.11	1096.61	1092.13	1087.66	1083.21	1078.77	1074.35	1069.94	1065.54	1061.16
18,000	1056.80	1052.44	1048.11	1043.79	1039.48	1035.18	1030.91	1026.64	1022.39	1018.16
19,000	1013.93	1009.73	1005.53	1001.35	997.188	993.036	988.899	984.776	980.667	976.571
20,000	972.490	968.422	964.368	960.328	956.302	952.289	948.290	944.304	940.332	936.374
21,000	932.429	928.497	924.579	920.675	916.783	912.906	909.041	905.190	901.351	897.527
22,000	893.715	889.916	886.130	882.358	878.598	874.852	871.118	867.398	863.690	859.995
23,000	856.313	852.643	848.986	845.343	841.711	838.093	834.487	830.893	827.312	823.744
24,000	820.188	816.645	813.114	809.595	806.089	802.595	799.113	795.643	792.186	788.741
25,000	785.308	781.887	778.478	775.081	771.697	768.324	764.963	761.614	758.277	754.952
26,000	751.638	748.337	745.047	741.769	738.502	735.248	732.005	728.773	725.553	722.345
27,000	719.148	715.962	712.788	709.625	706.474	703.334	700.206	697.088	693.982	690.887
28,000	687.803	684.731	681.670	678.619	675.580	672.552	669.535	666.528	663.533	660.548
29,000	657.575	654.612	651.660	648.719	645.789	642.869	639.960	637.062	634.175	631.298
30,000	628.431	625.575	622.730	619.895	617.071	614.257	611.453	608.660	605.877	603.105
31,000	600.342	597.590	594.849	592.117	589.396	586.684	583.983	581.292	578.611	575.940
32,000	573.279	570.628	567.987	565.355	562.734	560.123	557.521	554.929	552.347	549.775
33,000	547.212	544.659	542.115	539.582	537.058	534.543	532.038	529.543	527.057	524.580
34,000	522.113	519.655	517.207	514.768	512.338	509.918	507.507	505.105	502.712	500.329
35,000	497.955	495.589	493.233	490.887	488.549	486.220	483.900	481.589	479.287	476.994
36,000	474.710	472.434	470.169	467.914	465.671	463.439	461.216	459.001	456.794	454.596
37,000	452.434	450.268	448.106	445.947	443.791	441.640	439.495	437.357	435.226	433.101
38,000	431.203	429.096	427.077	425.061	423.048	421.038	419.031	417.027	415.026	413.028
39,000	410.968	408.911	407.036	405.164	403.294	401.427	399.563	397.702	395.844	393.989
40,000	391.682	389.675	387.935	386.194	384.454	382.716	380.980	379.247	377.517	375.789
41,000	373.302	371.345	369.731	368.117	366.504	364.893	363.284	361.677	360.072	358.469
42,000	355.784	353.877	352.381	350.886	349.392	347.900	346.409	344.920	343.432	341.945
43,000	339.088	337.231	335.845	334.460	333.076	331.693	330.311	328.931	327.552	326.174
44,000	323.176	321.369	320.084	318.800	317.517	316.235	314.954	313.674	312.395	311.117
45,000	308.011	306.254	305.064	303.875	302.687	301.500	300.314	299.129	297.945	296.762
46,000	293.557	291.850	290.748	289.647	288.547	287.448	286.350	285.253	284.157	283.062
47,000	279.781	278.124	277.105	276.087	275.070	274.054	273.039	272.025	271.012	270.000
48,000	266.652	265.045	264.101	263.157	262.214	261.272	260.331	259.391	258.452	257.514
49,000	254.139	252.572	251.708	250.744	249.781	248.819	247.858	246.898	245.939	244.981

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TABLE VI. - STATIC PRESSURE p IN POUNDS PER SQUARE FOOT FOR VALUES
 OF PRESSURE ALTITUDE H IN GEOPOTENTIAL FEET - Concluded
 [These values are from refs. 8 and 10]

Pressure altitude, H , geopotential ft	0	200	400	500	600	800
50,000	242.213	239.896	237.601		235.328	233.076
51,000	230.847	228.638	226.451		224.285	222.139
52,000	220.014	217.909	215.824		213.760	211.715
53,000	209.689	207.683	205.696		203.729	201.780
54,000	199.849	197.937	196.044		194.168	192.311
55,000	190.471	188.649	186.844		185.057	183.286
56,000	181.533	179.796	178.076		176.373	174.685
57,000	173.014	171.359	169.720		168.096	166.488
58,000	164.895	163.318	161.755		160.208	158.675
59,000	157.157	155.654	154.165		152.690	151.229
60,000	149.782	148.349	146.930		145.525	144.132
61,000	142.754	141.388	140.035		138.696	137.369
62,000	136.055	134.753	133.464		132.187	130.923
63,000	129.670	128.430	127.201		125.984	124.779
64,000	123.585	122.403	121.232		120.072	118.923
65,000	117.786	116.659	115.543		114.437	113.343
66,000	112.26			109.59		
67,000	106.99			104.45		
68,000	101.97			99.548		
69,000	97.184			94.877		
70,000	92.624			90.424		
71,000	88.277			86.181		
72,000	84.135			82.137		
73,000	80.187			78.282		
74,000	76.424			74.609		
75,000	72.837			71.108		
76,000	69.419			67.771		
77,000	66.162			64.591		
78,000	63.057			61.560		
79,000	60.098			58.671		
80,000	57.278			55.918		
81,000	54.590			53.294		
82,000	52.028			50.794		
83,000	49.591			48.420		
84,000	47.278			46.166		
85,000	45.082			44.026		
86,000	42.996			41.993		
87,000	41.015			40.062		
88,000	39.133			38.227		
89,000	37.344			36.484		
90,000	35.644			34.826		
91,000	34.028			33.250		
92,000	32.491			31.751		
93,000	31.030			30.326		
94,000	29.639			28.970		
95,000	28.317			27.679		
96,000	27.058			26.451		
97,000	25.859			25.282		
98,000	24.719			24.169		
99,000	23.632			23.109		
100,000	22.598					

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TABLE VII. - RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER

[These values are from ref. 3]

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Mach number, M	0	1	2	3	4	5	6	7	8	9
0.10	0.00702	0.00716	0.00730	0.00745	0.00759	0.00774	0.00789	0.00804	0.00819	0.00834
.11	.00850	.00865	.00881	.00897	.00913	.00929	.00945	.00962	.00978	.00995
.12	.01012	.01029	.01046	.01063	.01080	.01098	.01116	.01134	.01152	.01170
.13	.01188	.01206	.01225	.01244	.01263	.01282	.01301	.01320	.01339	.01359
.14	.01379	.01399	.01419	.01439	.01459	.01480	.01500	.01521	.01542	.01563
.15	.01584	.01605	.01627	.01648	.01670	.01692	.01714	.01736	.01758	.01781
.16	.01804	.01826	.01849	.01872	.01895	.01919	.01942	.01966	.01990	.02014
.17	.02038	.02062	.02086	.02111	.02135	.02160	.02185	.02210	.02236	.02261
.18	.02286	.02312	.02338	.02364	.02390	.02416	.02443	.02469	.02496	.02523
.19	.02550	.02577	.02604	.02632	.02659	.02687	.02715	.02743	.02771	.02800
.20	.02828	.02857	.02886	.02914	.02944	.02973	.03002	.03032	.03061	.03091
.21	.03121	.03151	.03182	.03212	.03243	.03273	.03304	.03335	.03366	.03398
.22	.03429	.03461	.03493	.03525	.03557	.03589	.03621	.03654	.03686	.03719
.23	.03752	.03785	.03819	.03852	.03886	.03919	.03953	.03987	.04022	.04056
.24	.04090	.04125	.04160	.04195	.04230	.04265	.04301	.04336	.04372	.04408
.25	.04444	.04480	.04516	.04553	.04589	.04626	.04663	.04700	.04738	.04775
.26	.04813	.04850	.04888	.04926	.04964	.05003	.05041	.05080	.05119	.05158
.27	.05197	.05236	.05275	.05315	.05355	.05395	.05435	.05475	.05515	.05556
.28	.05596	.05637	.05678	.05719	.05761	.05802	.05844	.05886	.05927	.05970
.29	.06012	.06054	.06097	.06140	.06182	.06225	.06269	.06312	.06356	.06399
.30	.06443	.06487	.06531	.06575	.06620	.06665	.06709	.06754	.06799	.06845
.31	.06890	.06936	.06982	.07027	.07074	.07120	.07166	.07213	.07259	.07306
.32	.07353	.07401	.07448	.07496	.07543	.07591	.07639	.07687	.07736	.07784
.33	.07833	.07882	.07931	.07980	.08029	.08079	.08128	.08178	.08228	.08278
.34	.08329	.08379	.08430	.08481	.08531	.08583	.08634	.08685	.08737	.08789
.35	.08841	.08893	.08945	.08998	.09050	.09103	.09156	.09209	.09263	.09316
.36	.09370	.09424	.09478	.09532	.09586	.09641	.09695	.09750	.09805	.09860
.37	.09916	.09971	.10027	.10083	.10139	.10195	.10251	.10308	.10364	.10421
.38	.10478	.10535	.10593	.10650	.10708	.10766	.10824	.10882	.10941	.10999
.39	.11058	.11117	.11176	.11235	.11295	.11354	.11414	.11474	.11534	.11595
.40	.11655	.11716	.11777	.11838	.11899	.11960	.12022	.12084	.12146	.12208
.41	.12270	.12332	.12395	.12458	.12521	.12584	.12647	.12711	.12774	.12838
.42	.12902	.12966	.13031	.13095	.13160	.13225	.13290	.13355	.13421	.13487
.43	.13552	.13618	.13685	.13751	.13818	.13884	.13951	.14018	.14086	.14153
.44	.14221	.14289	.14357	.14425	.14493	.14562	.14630	.14699	.14768	.14838
.45	.14907	.14977	.15047	.15117	.15187	.15257	.15328	.15399	.15470	.15541
.46	.15612	.15684	.15755	.15827	.15899	.15972	.16044	.16117	.16190	.16263
.47	.16336	.16409	.16483	.16557	.16631	.16705	.16779	.16854	.16928	.17003
.48	.17079	.17154	.17229	.17305	.17381	.17457	.17533	.17610	.17686	.17763
.49	.17840	.17917	.17995	.18072	.18150	.18228	.18307	.18385	.18463	.18542
.50	.18621	.18700	.18780	.18859	.18939	.19019	.19099	.19180	.19260	.19341
.51	.19422	.19503	.19584	.19666	.19748	.19830	.19912	.19994	.20077	.20159
.52	.20242	.20326	.20409	.20492	.20576	.20660	.20744	.20829	.20913	.20995
.53	.21083	.21168	.21253	.21339	.21425	.21511	.21597	.21683	.21770	.21857
.54	.21944	.22031	.22118	.22206	.22294	.22382	.22470	.22559	.22647	.22736
.55	.22825	.22914	.23004	.23094	.23184	.23274	.23364	.23455	.23545	.23636
.56	.23727	.23819	.23910	.24002	.24094	.24186	.24279	.24372	.24464	.24558
.57	.24651	.24744	.24838	.24932	.25026	.25121	.25215	.25310	.25405	.25500
.58	.25596	.25691	.25787	.25883	.25980	.26076	.26173	.26270	.26367	.26464
.59	.26562	.26660	.26758	.26856	.26955	.27053	.27152	.27252	.27351	.27451
.60	.27550	.27650	.27751	.27851	.27952	.28053	.28154	.28255	.28357	.28459
.61	.28561	.28663	.28766	.28869	.28972	.29075	.29178	.29282	.29386	.29490
.62	.29594	.29699	.29804	.29909	.30014	.30119	.30225	.30331	.30437	.30544
.63	.30650	.30757	.30864	.30972	.31079	.31187	.31295	.31403	.31512	.31621
.64	.31729	.31839	.31948	.32058	.32168	.32278	.32388	.32499	.32610	.32721
.65	.32832	.32944	.33056	.33168	.33280	.33393	.33505	.33618	.33732	.33845
.66	.33959	.34073	.34187	.34301	.34416	.34531	.34646	.34762	.34877	.34993
.67	.35110	.35226	.35343	.35460	.35577	.35694	.35812	.35930	.36048	.36166
.68	.36285	.36404	.36523	.36642	.36762	.36882	.37002	.37122	.37243	.37364
.69	.37485	.37606	.37728	.37850	.37972	.38094	.38217	.38340	.38463	.38586

TABLE VII.- RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER - Continued

[These values are from ref. 3]

Mach number, M	0	1	2	3	4	5	6	7	8	9
0.70	0.38710	0.38834	0.38958	0.39083	0.39207	0.39332	0.39458	0.39583	0.39709	0.39835
.71	.39961	.40088	.40214	.40341	.40469	.40596	.40724	.40852	.40980	.41109
.72	.41238	.41367	.41496	.41626	.41756	.41886	.42017	.42147	.42278	.42410
.73	.42541	.42673	.42805	.42937	.43070	.43203	.43336	.43469	.43603	.43737
.74	.43871	.44005	.44140	.44275	.44410	.44546	.44682	.44818	.44954	.45091
.75	.45228	.45365	.45503	.45640	.45778	.45917	.46055	.46194	.46333	.46473
.76	.46612	.46752	.46893	.47033	.47174	.47315	.47457	.47598	.47740	.47882
.77	.48025	.48168	.48311	.48454	.48598	.48742	.48886	.49030	.49175	.49320
.78	.49466	.49611	.49757	.49903	.50050	.50197	.50344	.50491	.50639	.50787
.79	.50935	.51084	.51233	.51382	.51531	.51681	.51831	.51981	.52132	.52283
.80	.52434	.52586	.52737	.52889	.53042	.53195	.53347	.53501	.53654	.53808
.81	.53962	.54117	.54272	.54427	.54582	.54738	.54894	.55050	.55207	.55364
.82	.55521	.55679	.55836	.55994	.56153	.56312	.56471	.56630	.56790	.56950
.83	.57110	.57271	.57432	.57593	.57754	.57916	.58078	.58241	.58404	.58567
.84	.58730	.58894	.59058	.59222	.59387	.59552	.59717	.59883	.60049	.60215
.85	.60382	.60549	.60716	.60884	.61051	.61220	.61388	.61557	.61726	.61896
.86	.62066	.62236	.62406	.62577	.62748	.62920	.63091	.63263	.63436	.63609
.87	.63782	.63955	.64129	.64303	.64477	.64652	.64827	.65003	.65178	.65354
.88	.65531	.65708	.65885	.66062	.66240	.66418	.66596	.66775	.66954	.67134
.89	.67314	.67494	.67674	.67855	.68036	.68218	.68399	.68582	.68764	.68947
.90	.69130	.69314	.69498	.69682	.69867	.70052	.70237	.70423	.70609	.70795
.91	.70982	.71169	.71356	.71544	.71732	.71920	.72109	.72298	.72488	.72678
.92	.72868	.73059	.73250	.73441	.73633	.73825	.74017	.74210	.74403	.74596
.93	.74790	.74984	.75179	.75374	.75569	.75765	.75961	.76157	.76354	.76551
.94	.76749	.76946	.77143	.77341	.77542	.77742	.77941	.78141	.78342	.78543
.95	.78744	.78945	.79147	.79350	.79552	.79755	.79959	.80163	.80367	.80571
.96	.80776	.80982	.81187	.81394	.81600	.81807	.82014	.82222	.82430	.82638
.97	.82847	.83056	.83266	.83476	.83686	.83897	.84108	.84319	.84531	.84744
.98	.84956	.85169	.85383	.85597	.85811	.86025	.86241	.86456	.86672	.86888
.99	.87105	.87322	.87539	.87757	.87975	.88194	.88413	.88632	.88852	.89072
1.00	.89293	.89514	.89735	.89957	.90180	.90402	.90625	.90849	.91073	.91297
1.01	.91521	.91746	.91972	.92198	.92424	.92651	.92878	.93105	.93333	.93561
1.02	.93790	.94019	.94248	.94478	.94708	.94938	.95169	.95401	.95632	.95864
1.03	.96097	.96330	.96563	.96796	.97030	.97265	.97500	.97735	.97970	.98206
1.04	.98442	.98679	.98916	.99153	.99391	.99629	.99868	1.00106	1.00346	1.00585
1.05	1.00825	1.01066	1.01306	1.01547	1.01789	1.02031	1.02273	1.02515	1.02758	1.03002
1.06	1.03245	1.03489	1.03734	1.03978	1.04224	1.04469	1.04715	1.04961	1.05208	1.05455
1.07	1.05702	1.05949	1.06197	1.06446	1.06694	1.06944	1.07193	1.07443	1.07693	1.07943
1.08	1.08194	1.08445	1.08697	1.08949	1.09201	1.09454	1.09707	1.09960	1.10214	1.10468
1.09	1.10722	1.10977	1.11232	1.11487	1.11743	1.11999	1.12255	1.12512	1.12769	1.13027
1.10	1.13285	1.13543	1.13801	1.14060	1.14320	1.14579	1.14839	1.15099	1.15360	1.15621
1.11	1.15882	1.16144	1.16406	1.16668	1.16930	1.17193	1.17457	1.17720	1.17984	1.18249
1.12	1.18513	1.18778	1.19044	1.19309	1.19575	1.19842	1.20108	1.20375	1.20643	1.20910
1.13	1.21178	1.21447	1.21715	1.21985	1.22254	1.22524	1.22794	1.23064	1.23335	1.23606
1.14	1.23877	1.24149	1.24421	1.24693	1.24966	1.25239	1.25512	1.25785	1.26059	1.26334
1.15	1.26608	1.26883	1.27159	1.27434	1.27710	1.27986	1.28263	1.28540	1.28817	1.29095
1.16	1.29372	1.29651	1.29929	1.30208	1.30487	1.30767	1.31047	1.31327	1.31607	1.31888
1.17	1.32169	1.32450	1.32732	1.33014	1.33297	1.33579	1.33862	1.34146	1.34429	1.34713
1.18	1.34998	1.35282	1.35567	1.35852	1.36138	1.36424	1.36710	1.36997	1.37284	1.37571
1.19	1.37858	1.38146	1.38434	1.38722	1.39011	1.39300	1.39590	1.39879	1.40169	1.40460
1.20	1.40750	1.41041	1.41332	1.41624	1.41916	1.42208	1.42500	1.42793	1.43086	1.43380
1.21	1.43674	1.43968	1.44262	1.44557	1.44852	1.45147	1.45442	1.45738	1.46033	1.46331
1.22	1.46628	1.46925	1.47223	1.47520	1.47818	1.48117	1.48416	1.48715	1.49014	1.49313
1.23	1.49613	1.49914	1.50214	1.50515	1.50816	1.51118	1.51419	1.51721	1.52024	1.52326
1.24	1.52629	1.52933	1.53236	1.53540	1.53844	1.54149	1.54454	1.54759	1.55064	1.55370
1.25	1.55676	1.55982	1.56289	1.56596	1.56903	1.57210	1.57518	1.57826	1.58135	1.58444
1.26	1.58753	1.59062	1.59372	1.59682	1.59992	1.60302	1.60612	1.60924	1.61236	1.61548
1.27	1.61860	1.62172	1.62485	1.62797	1.63111	1.63424	1.63738	1.64052	1.64367	1.64681
1.28	1.64996	1.65321	1.65647	1.65973	1.66300	1.66627	1.66953	1.67280	1.67607	1.67934
1.29	1.68163	1.68481	1.68800	1.69119	1.69438	1.69758	1.70077	1.70397	1.70718	1.71038

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TABLE VII. - RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER - Continued

[These values are from ref. 3]

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Mach number, M	0	1	2	3	4	5	6	7	8	9
1.30	1.71359	1.71681	1.72002	1.72324	1.72646	1.72969	1.73291	1.73614	1.73938	1.74261
1.31	1.74585	1.74909	1.75234	1.75559	1.75884	1.76209	1.76535	1.76861	1.77187	1.77513
1.32	1.77840	1.78167	1.78495	1.78823	1.79151	1.79479	1.79807	1.80136	1.80465	1.80795
1.33	1.81125	1.81455	1.81785	1.82116	1.82447	1.82778	1.83109	1.83441	1.83773	1.84105
1.34	1.84438	1.84771	1.85104	1.85438	1.85772	1.86106	1.86440	1.86775	1.87110	1.87445
1.35	1.87781	1.88116	1.88452	1.88789	1.89126	1.89463	1.89800	1.90137	1.90475	1.90813
1.36	1.91152	1.91491	1.91830	1.92169	1.92508	1.92848	1.93188	1.93529	1.93870	1.94211
1.37	1.94552	1.94893	1.95235	1.95577	1.95920	1.96263	1.96606	1.96949	1.97293	1.97636
1.38	1.97981	1.98325	1.98670	1.99015	1.99360	1.99706	2.00052	2.00398	2.00744	2.01091
1.39	2.01438	2.01785	2.02133	2.02481	2.02829	2.03177	2.03526	2.03875	2.04224	2.04574
1.40	2.04924	2.05274	2.05624	2.05975	2.06326	2.06677	2.07029	2.07380	2.07733	2.08085
1.41	2.08438	2.08791	2.09144	2.09497	2.09851	2.10205	2.10560	2.10914	2.11269	2.11624
1.42	2.11980	2.12336	2.12692	2.13048	2.13405	2.13762	2.14119	2.14476	2.14834	2.15192
1.43	2.15551	2.15909	2.16268	2.16627	2.16987	2.17346	2.17706	2.18067	2.18427	2.18788
1.44	2.19149	2.19511	2.19872	2.20234	2.20597	2.20959	2.21322	2.21685	2.22048	2.22412
1.45	2.22776	2.23140	2.23505	2.23869	2.24234	2.24599	2.24965	2.25331	2.25697	2.26064
1.46	2.26431	2.26798	2.27165	2.27532	2.27900	2.28268	2.28637	2.29005	2.29374	2.29744
1.47	2.30113	2.30483	2.30853	2.31223	2.31594	2.31965	2.32336	2.32707	2.33079	2.33451
1.48	2.33823	2.34196	2.34569	2.34942	2.35315	2.35689	2.36063	2.36437	2.36812	2.37187
1.49	2.37562	2.37937	2.38313	2.38688	2.39065	2.39441	2.39818	2.40195	2.40572	2.40950
1.50	2.41327	2.41706	2.42084	2.42463	2.42842	2.43221	2.43600	2.43980	2.44360	2.44740
1.51	2.45121	2.45502	2.45883	2.46264	2.46646	2.47028	2.47410	2.47793	2.48176	2.48559
1.52	2.48942	2.49326	2.49710	2.50094	2.50478	2.50863	2.51248	2.51633	2.52019	2.52405
1.53	2.52791	2.53177	2.53564	2.53951	2.54338	2.54725	2.55113	2.55501	2.55889	2.56278
1.54	2.56667	2.57056	2.57445	2.57835	2.58225	2.58615	2.59005	2.59396	2.59787	2.60179
1.55	2.60570	2.60962	2.61354	2.61747	2.62139	2.62532	2.62925	2.63319	2.63713	2.64107
1.56	2.64501	2.64896	2.65290	2.65686	2.66081	2.66477	2.66873	2.67269	2.67665	2.68062
1.57	2.68459	2.68856	2.69254	2.69652	2.70050	2.70449	2.70847	2.71246	2.71645	2.72045
1.58	2.72445	2.72845	2.73245	2.73646	2.74046	2.74448	2.74849	2.75251	2.75653	2.76055
1.59	2.76457	2.76860	2.77263	2.77666	2.78070	2.78474	2.78878	2.79282	2.79687	2.80092
1.60	2.80497	2.80903	2.81308	2.81714	2.82121	2.82527	2.82934	2.83341	2.83749	2.84156
1.61	2.84564	2.84972	2.85381	2.85790	2.86199	2.86608	2.87017	2.87427	2.87837	2.88248
1.62	2.88658	2.89069	2.89480	2.89892	2.90304	2.90716	2.91128	2.91540	2.91953	2.92366
1.63	2.92780	2.93193	2.93607	2.94021	2.94436	2.94850	2.95265	2.95681	2.96096	2.96512
1.64	2.96928	2.97344	2.97761	2.98178	2.98595	2.99012	2.99430	2.99848	3.00266	3.00684
1.65	3.01103	3.01522	3.01941	3.02361	3.02781	3.03201	3.03621	3.04042	3.04463	3.04884
1.66	3.05305	3.05727	3.06149	3.06571	3.06994	3.07417	3.07840	3.08263	3.08687	3.09110
1.67	3.09535	3.09959	3.10384	3.10809	3.11234	3.11659	3.12085	3.12511	3.12937	3.13364
1.68	3.13791	3.14218	3.14645	3.15073	3.15501	3.15929	3.16357	3.16786	3.17215	3.17644
1.69	3.18074	3.18503	3.18933	3.19364	3.19794	3.20225	3.20656	3.21088	3.21519	3.21951
1.70	3.22383	3.22816	3.23248	3.23681	3.24115	3.24548	3.24982	3.25416	3.25850	3.26285
1.71	3.26720	3.27155	3.27590	3.28026	3.28462	3.28898	3.29335	3.29771	3.30208	3.30646
1.72	3.31083	3.31521	3.31959	3.32397	3.32836	3.33275	3.33714	3.34154	3.34593	3.35033
1.73	3.35473	3.35914	3.36355	3.36796	3.37237	3.37679	3.38120	3.38562	3.39005	3.39447
1.74	3.39890	3.40333	3.40777	3.41221	3.41665	3.42109	3.42553	3.42998	3.43443	3.43888
1.75	3.44334	3.44780	3.45226	3.45672	3.46119	3.46566	3.47013	3.47460	3.47908	3.48356
1.76	3.48804	3.49253	3.49701	3.50150	3.50600	3.51049	3.51499	3.51949	3.52400	3.52850
1.77	3.53301	3.53752	3.54204	3.54655	3.55107	3.55560	3.56012	3.56465	3.56918	3.57371
1.78	3.57825	3.58278	3.58733	3.59187	3.59642	3.60096	3.60552	3.61007	3.61463	3.61919
1.79	3.62375	3.62831	3.63288	3.63745	3.64202	3.64660	3.65118	3.65576	3.66034	3.66493
1.80	3.66952	3.67411	3.67870	3.68330	3.68790	3.69250	3.69710	3.70171	3.70632	3.71093
1.81	3.71555	3.72017	3.72479	3.72941	3.73404	3.73867	3.74330	3.74793	3.75257	3.75721
1.82	3.76185	3.76649	3.77114	3.77579	3.78044	3.78510	3.78975	3.79442	3.79908	3.80374
1.83	3.80841	3.81308	3.81776	3.82243	3.82711	3.83179	3.83648	3.84117	3.84585	3.85055
1.84	3.85524	3.85994	3.86464	3.86934	3.87405	3.87876	3.88347	3.88818	3.89290	3.89761
1.85	3.90234	3.90706	3.91179	3.91652	3.92125	3.92598	3.93072	3.93546	3.94020	3.94495
1.86	3.94970	3.95445	3.95920	3.96396	3.96871	3.97347	3.97822	3.98300	3.98777	3.99255
1.87	3.99732	4.00210	4.00688	4.01166	4.01644	4.02123	4.02602	4.03081	4.03561	4.04041
1.88	4.04521	4.05001	4.05482	4.05963	4.06444	4.06925	4.07407	4.07889	4.08371	4.08853
1.89	4.09336	4.09819	4.10302	4.10786	4.11270	4.11754	4.12238	4.12722	4.13207	4.13692

TABLE VII. - RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER - Continued

[These values are from ref. 3]

Mach number, M	0	1	2	3	4	5	6	7	8	9
1.90	4.14178	4.14663	4.15149	4.15635	4.16122	4.16608	4.17095	4.17583	4.18070	4.18558
1.91	4.19046	4.19534	4.20023	4.20511	4.21000	4.21490	4.21979	4.22469	4.22959	4.23450
1.92	4.23940	4.24431	4.24922	4.25414	4.25907	4.26397	4.26890	4.27382	4.27875	4.28368
1.93	4.28861	4.29355	4.29848	4.30342	4.30837	4.31331	4.31826	4.32321	4.32817	4.33312
1.94	4.33808	4.34304	4.34801	4.35298	4.35795	4.36292	4.36789	4.37287	4.37785	4.38283
1.95	4.38782	4.39281	4.39780	4.40279	4.40779	4.41278	4.41779	4.42278	4.42780	4.43280
1.96	4.43782	4.44283	4.44785	4.45287	4.45789	4.46291	4.46794	4.47297	4.47800	4.48304
1.97	4.48308	4.48812	4.49316	4.50321	4.50826	4.51331	4.51836	4.52342	4.52848	4.53354
1.98	4.53860	4.54367	4.54874	4.55381	4.55889	4.56396	4.56904	4.57413	4.57921	4.58430
1.99	4.58939	4.59448	4.59958	4.60468	4.60978	4.61488	4.61999	4.62510	4.63021	4.63532
2.00	4.64044	4.64556	4.65068	4.65581	4.66093	4.66606	4.67120	4.67633	4.68147	4.68661
2.01	4.69175	4.69690	4.70205	4.70720	4.71235	4.71751	4.72267	4.72783	4.73299	4.73816
2.02	4.74333	4.74850	4.75368	4.75881	4.76405	4.76922	4.77440	4.77959	4.78478	4.78997
2.03	4.79517	4.80037	4.80557	4.81077	4.81598	4.82119	4.82640	4.83161	4.83683	4.84205
2.04	4.84727	4.85249	4.85772	4.86295	4.86818	4.87342	4.87865	4.88388	4.88914	4.89438
2.05	4.89963	4.90488	4.91014	4.91539	4.92065	4.92591	4.93117	4.93644	4.94171	4.94698
2.06	4.95226	4.95753	4.96281	4.96809	4.97338	4.97867	4.98396	4.98925	4.99454	4.99984
2.07	5.00514	5.01045	5.01575	5.02106	5.02637	5.03168	5.03700	5.04232	5.04764	5.05296
2.08	5.05829	5.06362	5.06895	5.07429	5.07962	5.08496	5.09031	5.09565	5.10100	5.10635
2.09	5.11170	5.11706	5.12242	5.12778	5.13314	5.13851	5.14387	5.14923	5.15460	5.16000
2.10	5.16538	5.17076	5.17614	5.18153	5.18692	5.19231	5.19770	5.20310	5.20850	5.21390
2.11	5.21931	5.22472	5.23013	5.23554	5.24096	5.24637	5.25180	5.25722	5.26265	5.26807
2.12	5.27351	5.27894	5.28438	5.28981	5.29526	5.30070	5.30615	5.31160	5.31705	5.32250
2.13	5.32796	5.33342	5.33889	5.34435	5.34982	5.35529	5.36076	5.36624	5.37172	5.37720
2.14	5.38268	5.38817	5.39366	5.39915	5.40464	5.41014	5.41564	5.42114	5.42664	5.43215
2.15	5.43766	5.44317	5.44869	5.45421	5.45973	5.46525	5.47077	5.47630	5.48182	5.48737
2.16	5.49290	5.49844	5.50398	5.50953	5.51507	5.52062	5.52617	5.53173	5.53728	5.54284
2.17	5.54841	5.55397	5.55954	5.56511	5.57068	5.57625	5.58183	5.58741	5.59300	5.59858
2.18	5.60417	5.60976	5.61535	5.62095	5.62655	5.63215	5.63775	5.64336	5.64897	5.65458
2.19	5.66019	5.66581	5.67143	5.67705	5.68268	5.68830	5.69393	5.69957	5.70520	5.71084
2.20	5.71648	5.72212	5.72777	5.73342	5.73907	5.74472	5.75038	5.75604	5.76170	5.76736
2.21	5.77303	5.77870	5.78437	5.79004	5.79572	5.80140	5.80708	5.81276	5.81845	5.82414
2.22	5.82983	5.83553	5.84123	5.84693	5.85263	5.85834	5.86404	5.86976	5.87548	5.88118
2.23	5.88690	5.89262	5.89835	5.90407	5.90980	5.91554	5.92127	5.92701	5.93275	5.93849
2.24	5.94423	5.94998	5.95573	5.96148	5.96724	5.97299	5.97875	5.98452	5.99028	5.99605
2.25	6.00182	6.00760	6.01337	6.01915	6.02493	6.03071	6.03650	6.04229	6.04808	6.05388
2.26	6.05967	6.06547	6.07127	6.07707	6.08289	6.08870	6.09451	6.10032	6.10614	6.11196
2.27	6.11778	6.12361	6.12944	6.13527	6.14110	6.14694	6.15278	6.15862	6.16446	6.17031
2.28	6.17616	6.18201	6.18786	6.19372	6.19958	6.20544	6.21130	6.21717	6.22304	6.22891
2.29	6.23479	6.24066	6.24654	6.25243	6.25831	6.26420	6.27009	6.27598	6.28188	6.28778
2.30	6.29368	6.29958	6.30549	6.31140	6.31731	6.32322	6.32914	6.33506	6.34098	6.34691
2.31	6.35283	6.35876	6.36469	6.37063	6.37657	6.38251	6.38845	6.39439	6.40034	6.40629
2.32	6.41225	6.41820	6.42416	6.43012	6.43608	6.44205	6.44802	6.45399	6.45996	6.46594
2.33	6.47192	6.47790	6.48388	6.48987	6.49586	6.50185	6.50784	6.51384	6.51984	6.52585
2.34	6.53185	6.53786	6.54387	6.54988	6.55590	6.56192	6.56794	6.57396	6.57999	6.58601
2.35	6.59205	6.59808	6.60412	6.61015	6.61620	6.62224	6.62829	6.63434	6.64039	6.64644
2.36	6.65250	6.65856	6.66462	6.67069	6.67675	6.68282	6.68889	6.69497	6.70105	6.70713
2.37	6.71321	6.71930	6.72539	6.73148	6.73757	6.74367	6.74977	6.75587	6.76197	6.76808
2.38	6.77419	6.78030	6.78641	6.79253	6.79865	6.80477	6.81089	6.81702	6.82315	6.82929
2.39	6.83542	6.84156	6.84770	6.85384	6.85999	6.86613	6.87229	6.87844	6.88459	6.89075
2.40	6.89691	6.90308	6.90924	6.91541	6.92158	6.92776	6.93395	6.94011	6.94630	6.95248
2.41	6.95867	6.96486	6.97105	6.97724	6.98344	6.98964	6.99584	7.00205	7.00826	7.01447
2.42	7.02068	7.02690	7.03311	7.03932	7.04556	7.05178	7.05801	7.06424	7.07048	7.07672
2.43	7.08295	7.08920	7.09544	7.10169	7.10794	7.11419	7.12044	7.12670	7.13296	7.13922
2.44	7.14549	7.15175	7.15802	7.16429	7.17057	7.17685	7.18313	7.18941	7.19570	7.20199
2.45	7.20828	7.21457	7.22087	7.22717	7.23347	7.23977	7.24608	7.25239	7.25870	7.26501
2.46	7.27133	7.27765	7.28397	7.29030	7.29663	7.30296	7.30929	7.31562	7.32196	7.32830
2.47	7.33464	7.34099	7.34734	7.35369	7.36004	7.36640	7.37275	7.37912	7.38548	7.39185
2.48	7.39821	7.40459	7.41096	7.41734	7.42372	7.43010	7.43648	7.44287	7.44926	7.45565
2.49	7.46205	7.46844	7.47484	7.48125	7.48765	7.49406	7.50047	7.50688	7.51330	7.51972

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TABLE VII. - RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER - Continued

[These values are from ref. 3]

Mach number, M	0	1	2	3	4	5	6	7	8	9
2.50	7.52614	7.53256	7.53899	7.54541	7.55184	7.55828	7.56471	7.57115	7.57760	7.58404
2.51	7.59049	7.59694	7.60339	7.60984	7.61630	7.62276	7.62922	7.63568	7.64215	7.64862
2.52	7.65510	7.66157	7.66805	7.67453	7.68101	7.68750	7.69399	7.70048	7.70697	7.71347
2.53	7.71996	7.72647	7.73297	7.73948	7.74598	7.75250	7.75901	7.76553	7.77205	7.77857
2.54	7.78509	7.79162	7.79815	7.80468	7.81122	7.81775	7.82429	7.83084	7.83738	7.84393
2.55	7.85048	7.85703	7.86359	7.87015	7.87671	7.88327	7.88984	7.89641	7.90298	7.90955
2.56	7.91613	7.92271	7.92929	7.93587	7.94246	7.94905	7.95564	7.96223	7.96883	7.97543
2.57	7.98203	7.98864	7.99525	8.00186	8.00847	8.01508	8.02170	8.02832	8.03494	8.04157
2.58	8.04820	8.05483	8.06146	8.06810	8.07474	8.08138	8.08802	8.09467	8.10132	8.10797
2.59	8.11462	8.12128	8.12794	8.13460	8.14127	8.14793	8.15460	8.16128	8.16795	8.17463
2.60	8.18131	8.18799	8.19468	8.20136	8.20805	8.21475	8.22144	8.22814	8.23484	8.24154
2.61	8.24825	8.25496	8.26167	8.26838	8.27510	8.28182	8.28854	8.29527	8.30201	8.30872
2.62	8.31545	8.32219	8.32892	8.33566	8.34241	8.34915	8.35590	8.36265	8.36940	8.37616
2.63	8.38291	8.38968	8.39644	8.40320	8.40997	8.41674	8.42352	8.43029	8.43707	8.44385
2.64	8.45064	8.45742	8.46421	8.47100	8.47780	8.48459	8.49139	8.49819	8.50500	8.51181
2.65	8.51862	8.52543	8.53224	8.53906	8.54588	8.55270	8.55953	8.56636	8.57319	8.58002
2.66	8.58685	8.59369	8.60053	8.60738	8.61422	8.62107	8.62792	8.63478	8.64163	8.64849
2.67	8.65535	8.66222	8.66908	8.67595	8.68282	8.68970	8.69657	8.70345	8.71034	8.71722
2.68	8.72411	8.73100	8.73789	8.74479	8.75168	8.75858	8.76549	8.77239	8.77930	8.78621
2.69	8.79312	8.80004	8.80696	8.81388	8.82080	8.82773	8.83466	8.84159	8.84852	8.85546
2.70	8.86240	8.86934	8.87629	8.88323	8.89018	8.89713	8.90409	8.91105	8.91801	8.92497
2.71	8.93193	8.93890	8.94587	8.95284	8.95982	8.96680	8.97378	8.98076	8.98775	8.99473
2.72	9.00173	9.00872	9.01572	9.02271	9.02971	9.03672	9.04373	9.05073	9.05775	9.06476
2.73	9.07178	9.07880	9.08582	9.09284	9.09987	9.10690	9.11393	9.12097	9.12800	9.13505
2.74	9.14209	9.14913	9.15618	9.16323	9.17028	9.17734	9.18440	9.19146	9.19852	9.20559
2.75	9.21266	9.21973	9.22680	9.23388	9.24096	9.24804	9.25512	9.26221	9.26930	9.27639
2.76	9.28548	9.29258	9.29968	9.30678	9.31389	9.32100	9.32811	9.33522	9.34233	9.34945
2.77	9.35457	9.36169	9.36882	9.37595	9.38308	9.39021	9.39735	9.40449	9.41163	9.41877
2.78	9.42592	9.43307	9.44022	9.44737	9.45453	9.46169	9.46885	9.47601	9.48318	9.49035
2.79	9.49752	9.50470	9.51187	9.51905	9.52624	9.53342	9.54061	9.54780	9.55499	9.56219
2.80	9.56939	9.57659	9.58379	9.59099	9.59820	9.60541	9.61263	9.61984	9.62706	9.63428
2.81	9.64151	9.64873	9.65596	9.66319	9.67043	9.67767	9.68490	9.69215	9.69939	9.70664
2.82	9.71389	9.72114	9.72840	9.73565	9.74291	9.75018	9.75744	9.76471	9.77198	9.77925
2.83	9.78653	9.79381	9.80109	9.80837	9.81566	9.82294	9.83024	9.83753	9.84483	9.85212
2.84	9.85943	9.86673	9.87404	9.88135	9.88866	9.89597	9.90329	9.91061	9.91793	9.92526
2.85	9.93258	9.93991	9.94725	9.95458	9.96192	9.96926	9.97660	9.98395	9.99129	9.99865
2.86	10.00600	10.01335	10.02071	10.02807	10.03544	10.04280	10.05017	10.05754	10.06492	10.07229
2.87	10.07967	10.08705	10.09444	10.10183	10.10921	10.11661	10.12400	10.13140	10.13880	10.14620
2.88	10.15361	10.16101	10.16842	10.17584	10.18325	10.19067	10.19809	10.20551	10.21294	10.22037
2.89	10.22780	10.23523	10.24267	10.25010	10.25755	10.26499	10.27244	10.27988	10.28734	10.29479
2.90	10.30225	10.30971	10.31717	10.32463	10.33210	10.33957	10.34704	10.35452	10.36199	10.36947
2.91	10.37695	10.38444	10.39193	10.39942	10.40691	10.41441	10.42190	10.42940	10.43691	10.44441
2.92	10.45192	10.45943	10.46695	10.47446	10.48198	10.48950	10.49703	10.50455	10.51208	10.51961
2.93	10.52715	10.53468	10.54222	10.54977	10.55731	10.56486	10.57241	10.57996	10.58751	10.59507
2.94	10.60263	10.61019	10.61776	10.62533	10.63290	10.64047	10.64805	10.65562	10.66321	10.67079
2.95	10.67837	10.68596	10.69355	10.70115	10.70874	10.71634	10.72394	10.73155	10.73915	10.74676
2.96	10.75438	10.76199	10.76961	10.77723	10.78485	10.79247	10.80010	10.80773	10.81536	10.82300
2.97	10.83064	10.83828	10.84592	10.85356	10.86121	10.86886	10.87651	10.88417	10.89183	10.89949
2.98	10.90715	10.91482	10.92249	10.93016	10.93783	10.94551	10.95319	10.96087	10.96855	10.97624
2.99	10.98393	10.99162	10.99932	11.00701	11.01471	11.02241	11.03012	11.03783	11.04554	11.05325
3.00	11.06096	11.06868	11.07640	11.08413	11.09185	11.09958	11.10731	11.11504	11.12278	11.13052
3.01	11.13826	11.14600	11.15375	11.16150	11.16925	11.17700	11.18476	11.19252	11.20028	11.20804
3.02	11.21581	11.22358	11.23135	11.23913	11.24690	11.25468	11.26246	11.27025	11.27804	11.28583
3.03	11.29362	11.30142	11.30921	11.31701	11.32482	11.33262	11.34043	11.34824	11.35605	11.36387
3.04	11.37169	11.37951	11.38733	11.39516	11.40299	11.41082	11.41865	11.42649	11.43433	11.44217
3.05	11.45002	11.45786	11.46571	11.47356	11.48142	11.48928	11.49714	11.50500	11.51286	11.52073
3.06	11.52860	11.53647	11.54435	11.55223	11.56011	11.56799	11.57588	11.58377	11.59166	11.59955
3.07	11.60745	11.61534	11.62325	11.63115	11.63906	11.64696	11.65488	11.66279	11.67071	11.67863
3.08	11.68655	11.69447	11.70240	11.71033	11.71826	11.72620	11.73413	11.74207	11.75002	11.75796
3.09	11.76591	11.77386	11.78181	11.78977	11.79772	11.80569	11.81365	11.82161	11.82958	11.83755

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TABLE VII. - RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER - Continued

[These values are from ref. 3]

Mach number, M	0	1	2	3	4	5	6	7	8	9
3.70	17.09507	17.10459	17.11412	17.12365	17.13318	17.14271	17.15225	17.16179	17.17133	17.18088
3.71	17.19043	17.19998	17.20953	17.21909	17.22864	17.23821	17.24777	17.25733	17.26690	17.27647
3.72	17.28605	17.29562	17.30520	17.31478	17.32437	17.33395	17.34354	17.35313	17.36273	17.37233
3.73	17.38192	17.39153	17.40113	17.41074	17.42035	17.42996	17.43957	17.44919	17.45881	17.46843
3.74	17.47806	17.48769	17.49732	17.50695	17.51659	17.52622	17.53586	17.54551	17.55515	17.56480
3.75	17.57445	17.58411	17.59376	17.60342	17.61308	17.62275	17.63241	17.64208	17.65175	17.66143
3.76	17.67110	17.68078	17.69046	17.70015	17.70984	17.71953	17.72922	17.73891	17.74861	17.75831
3.77	17.76801	17.77772	17.78743	17.79714	17.80685	17.81656	17.82628	17.83600	17.84573	17.85545
3.78	17.86518	17.87491	17.88464	17.89438	17.90412	17.91386	17.92360	17.93335	17.94310	17.95285
3.79	17.96260	17.97236	17.98212	17.99188	18.00165	18.01141	18.02118	18.03095	18.04073	18.05051
3.80	18.06029	18.07007	18.07985	18.08964	18.09943	18.10922	18.11902	18.12882	18.13862	18.14842
3.81	18.15823	18.16804	18.17784	18.18766	18.19747	18.20729	18.21712	18.22693	18.23679	18.24659
3.82	18.25442	18.26426	18.27410	18.28393	18.29378	18.30362	18.31347	18.32332	18.33317	18.34302
3.83	18.35488	18.36474	18.37460	18.38447	18.39434	18.40421	18.41408	18.42395	18.43383	18.44371
3.84	18.45360	18.46348	18.47337	18.48326	18.49315	18.50305	18.51295	18.52285	18.53275	18.54266
3.85	18.55257	18.56248	18.57239	18.58231	18.59223	18.60215	18.61207	18.62200	18.63193	18.64186
3.86	18.65180	18.66173	18.67167	18.68161	18.69156	18.70151	18.71146	18.72141	18.73136	18.74132
3.87	18.75128	18.76125	18.77121	18.78118	18.79115	18.80112	18.81110	18.82108	18.83106	18.84104
3.88	18.85103	18.86102	18.87101	18.88100	18.89100	18.90100	18.91100	18.92100	18.93101	18.94102
3.89	18.95103	18.96105	18.97106	18.98108	18.99111	19.00113	19.01116	19.02119	19.03122	19.04125
3.90	19.05129	19.06133	19.07137	19.08142	19.09147	19.10152	19.11157	19.12163	19.13169	19.14175
3.91	19.15181	19.16188	19.17195	19.18202	19.19209	19.20217	19.21225	19.22233	19.23241	19.24250
3.92	19.25259	19.26268	19.27277	19.28287	19.29297	19.30307	19.31318	19.32328	19.33339	19.34351
3.93	19.35362	19.36374	19.37386	19.38398	19.39411	19.40424	19.41437	19.42450	19.43463	19.44477
3.94	19.45491	19.46506	19.47520	19.48535	19.49550	19.50566	19.51581	19.52597	19.53613	19.54630
3.95	19.55646	19.56663	19.57680	19.58698	19.59716	19.60733	19.61752	19.62770	19.63789	19.64808
3.96	19.65827	19.66847	19.67866	19.68886	19.69907	19.70927	19.71948	19.72969	19.73990	19.75012
3.97	19.76034	19.77056	19.78078	19.79101	19.80124	19.81147	19.82170	19.83194	19.84218	19.85242
3.98	19.86266	19.87291	19.88316	19.89341	19.90366	19.91392	19.92418	19.93444	19.94470	19.95497
3.99	19.96524	19.97551	19.98579	19.99607	20.00635	20.01663	20.02691	20.03720	20.04749	20.05779
4.00	20.06808	20.07838	20.08868	20.09898	20.10929	20.11960	20.12991	20.14022	20.15054	20.16086
4.01	20.17118	20.18150	20.19183	20.20216	20.21249	20.22282	20.23315	20.24350	20.25384	20.26419
4.02	20.27453	20.28488	20.29523	20.30559	20.31595	20.32631	20.33667	20.34703	20.35740	20.36777
4.03	20.37815	20.38852	20.39890	20.40928	20.41966	20.43005	20.44044	20.45083	20.46122	20.47162
4.04	20.48201	20.49242	20.50282	20.51323	20.52364	20.53405	20.54446	20.55488	20.56530	20.57572
4.05	20.58614	20.59657	20.60700	20.61743	20.62787	20.63830	20.64874	20.65919	20.66963	20.68008
4.06	20.69053	20.70098	20.71144	20.72189	20.73236	20.74282	20.75328	20.76375	20.77422	20.78470
4.07	20.79517	20.80565	20.81613	20.82662	20.83710	20.84759	20.85808	20.86858	20.87907	20.88957
4.08	20.90007	20.91058	20.92108	20.93159	20.94211	20.95262	20.96314	20.97366	20.98418	20.99470
4.09	21.00523	21.01576	21.02629	21.03683	21.04737	21.05791	21.06845	21.07900	21.08954	21.10009
4.10	21.11065	21.12121	21.13176	21.14232	21.15289	21.16345	21.17402	21.18459	21.19517	21.20575
4.11	21.21632	21.22690	21.23749	21.24808	21.25867	21.26926	21.27985	21.29045	21.30105	21.31165
4.12	21.32226	21.33286	21.34347	21.35409	21.36470	21.37532	21.38594	21.39656	21.40719	21.41782
4.13	21.42845	21.43908	21.44972	21.46035	21.47099	21.48164	21.49228	21.50293	21.51358	21.52424
4.14	21.53489	21.54555	21.55621	21.56688	21.57755	21.58822	21.59889	21.60956	21.62024	21.63092
4.15	21.64160	21.65228	21.66297	21.67366	21.68435	21.69505	21.70575	21.71645	21.72715	21.73786
4.16	21.74856	21.75927	21.76999	21.78070	21.79142	21.80214	21.81286	21.82359	21.83432	21.84505
4.17	21.85578	21.86652	21.87726	21.88800	21.89874	21.90949	21.92024	21.93099	21.94175	21.95250
4.18	21.96326	21.97402	21.98479	21.99556	22.00633	22.01710	22.02787	22.03865	22.04943	22.06021
4.19	22.07100	22.08179	22.09258	22.10337	22.11417	22.12496	22.13577	22.14657	22.15737	22.16818
4.20	22.17899	22.18981	22.20062	22.21144	22.22226	22.23309	22.24391	22.25474	22.26557	22.27641
4.21	22.28725	22.29808	22.30893	22.31977	22.33062	22.34147	22.35232	22.36317	22.37403	22.38489
4.22	22.39576	22.40662	22.41749	22.42836	22.43923	22.45011	22.46099	22.47186	22.48275	22.49363
4.23	22.50452	22.51541	22.52631	22.53720	22.54810	22.55900	22.56991	22.58081	22.59172	22.60263
4.24	22.61355	22.62446	22.63538	22.64631	22.65725	22.66816	22.67909	22.69002	22.70095	22.71189
4.25	22.72283	22.73377	22.74472	22.75567	22.76662	22.77757	22.78852	22.79948	22.81044	22.82141
4.26	22.83237	22.84334	22.85431	22.86528	22.87626	22.88724	22.89822	22.90921	22.92020	22.93118
4.27	22.94217	22.95317	22.96416	22.97516	22.98616	22.99717	23.00818	23.01918	23.03019	23.04121
4.28	23.05223	23.06325	23.07427	23.08529	23.09632	23.10735	23.11838	23.12942	23.14046	23.15150
4.29	23.16254	23.17359	23.18463	23.19569	23.20674	23.21779	23.22885	23.23991	23.25098	23.26204

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TABLE VII. - RATIO OF IMPACT PRESSURE TO STATIC PRESSURE q_c/p FOR VALUES OF MACH NUMBER - Concluded

[These values are from ref. 5]

Mach number, M	0	1	2	3	4	5	6	7	8	9
4.90	30.37898	30.39160	30.40422	30.41684	30.42947	30.44210	30.45473	30.46736	30.47999	30.49263
4.91	30.50527	30.51792	30.53056	30.54321	30.55586	30.56851	30.58117	30.59383	30.60649	30.61915
4.92	30.63182	30.64449	30.65716	30.66984	30.68251	30.69519	30.70787	30.72056	30.73324	30.74593
4.93	30.75863	30.77132	30.78402	30.79672	30.80942	30.82212	30.83483	30.84754	30.86025	30.87297
4.94	30.88569	30.89841	30.91113	30.92386	30.93659	30.94932	30.96205	30.97478	30.98752	31.00027
4.95	31.01301	31.02576	31.03850	31.05126	31.06401	31.07677	31.08952	31.10229	31.11505	31.12782
4.96	31.14059	31.15336	31.16613	31.17891	31.19169	31.20447	31.21726	31.23004	31.24284	31.25563
4.97	31.26842	31.28122	31.29402	31.30682	31.31963	31.33244	31.34525	31.35806	31.37088	31.38369
4.98	31.39651	31.40934	31.42217	31.43499	31.44782	31.46066	31.47349	31.48633	31.49917	31.51202
4.99	31.52487	31.53771	31.55057	31.56342	31.57628	31.58914	31.60200	31.61486	31.62773	31.64060
5.00	31.65347									

TABLE VIII. - SPEED OF SOUND • FOR VALUES OF FREE-AIR TEMPERATURE t IN DEGREES FAHRENHEIT

Temp., t, °F	0	1	2	3	4	5	6	7	8	9	
	Speed of sound, mph										
-70	659.798	667.373	666.536	665.697	664.858	664.017	663.175	662.333	661.489	660.644	
-60	668.210	675.691	674.864	674.036	673.207	672.376	671.545	670.713	669.880	669.045	
-50	676.517	683.998	683.171	682.342	681.513	680.683	679.852	679.020	678.187	677.353	
-40	684.724	692.205	691.378	690.549	689.719	688.889	688.058	687.226	686.393	685.559	
-30	692.834	700.315	699.488	698.659	697.829	696.998	696.167	695.335	694.502	693.669	
-20	700.849	708.330	707.503	706.673	705.843	705.012	704.181	703.350	702.519	701.686	
-10	708.774	716.255	715.428	714.598	713.767	712.936	712.105	711.274	710.443	709.612	
0	716.612	724.093	723.266	722.436	721.605	720.774	720.943	720.112	719.281	718.450	
10	724.364	731.845	731.018	730.188	729.357	728.526	727.695	726.864	726.033	725.202	
20	732.035	739.516	738.689	737.859	737.028	736.197	735.366	734.535	733.704	732.873	
30	739.626	747.107	746.280	745.450	744.620	743.789	742.958	742.127	741.296	740.465	
40	747.140	754.621	753.794	752.964	752.133	751.303	750.472	749.641	748.810	747.979	
50	754.579	762.060	761.233	760.403	759.572	758.742	757.911	757.080	756.249	755.418	
60	761.945	769.426	768.599	767.769	766.938	766.107	765.276	764.445	763.614	762.783	
70	769.241	776.722	775.895	775.064	774.233	773.402	772.571	771.740	770.909	770.078	
80	776.468	783.949	783.122	782.291	781.460	780.629	779.798	778.967	778.136	777.305	
90	783.629	791.110	790.283	789.452	788.621	787.790	786.959	786.128	785.297	784.466	
100	790.725	798.206	797.379	796.548	795.717	794.886	794.055	793.224	792.393	791.562	
110	797.757	805.238	804.411	803.580	802.749	801.918	801.087	800.256	799.425	798.594	
120	804.729										
	Speed of sound, knots										
-70	573.349	579.932	579.204	578.475	577.746	577.015	576.284	575.551	574.818	574.084	
-60	580.658	587.160	586.441	585.712	584.983	584.254	583.523	582.792	582.061	581.330	
-50	587.877	594.300	593.589	592.878	592.167	591.454	590.743	590.032	589.321	588.610	
-40	595.009	601.355	600.653	599.950	599.247	598.543	597.838	597.133	596.428	595.723	
-30	602.056	608.328	607.635	606.940	606.245	605.549	604.852	604.155	603.458	602.761	
-20	609.021	615.223	614.537	613.850	613.163	612.474	611.785	611.095	610.405	609.713	
-10	615.908	622.041	621.362	620.683	620.003	619.323	618.641	617.959	617.276	616.592	
0	622.718	628.795	628.072	627.347	626.622	625.896	625.169	624.442	623.714	622.985	
10	629.455	635.455	634.729	634.002	633.275	632.548	631.821	631.094	630.367	629.640	
20	636.121	642.041	641.314	640.587	639.860	639.133	638.406	637.679	636.952	636.225	
30	642.717	648.557	647.830	647.103	646.376	645.649	644.922	644.195	643.468	642.741	
40	649.247	655.007	654.280	653.553	652.826	652.099	651.372	650.645	649.918	649.191	
50	655.711	661.391	660.664	659.937	659.210	658.483	657.756	657.029	656.302	655.575	
60	662.112	667.712	666.985	666.258	665.531	664.804	664.077	663.350	662.623	661.896	
70	668.452	673.972	673.245	672.518	671.791	671.064	670.337	669.610	668.883	668.156	
80	674.732	680.172	679.445	678.718	677.991	677.264	676.537	675.810	675.083	674.356	
90	680.955	686.315	685.588	684.861	684.134	683.407	682.680	681.953	681.226	680.500	
100	687.121	692.341	691.614	690.887	690.160	689.433	688.706	687.979	687.252	686.525	
110	693.232										
120	699.290										

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TABLE IX.- SPEED OF SOUND a FOR VALUES OF FREE-AIR TEMPERATURE t IN DEGREES CENTIGRADE

Temp., t, °C	Speed of sound, mph									
	0	1	2	3	4	5	6	7	8	9
-60	654.699	668.377	666.871	665.362	663.849	662.333	660.813	659.290	657.763	656.233
-50	669.880	683.254	681.761	680.305	678.825	677.342	675.856	674.367	672.875	671.379
-40	684.724	697.814	696.372	694.927	693.478	692.027	690.573	689.115	687.654	686.191
-30	699.253	712.077	710.663	709.247	707.828	706.406	704.981	703.554	702.123	700.690
-20	713.487	724.673	724.673	723.284	721.893	720.498	719.102	717.702	716.300	714.895
-10	727.442	739.777	738.416	737.054	735.688	734.320	732.950	731.577	730.201	728.823
0	741.135	742.490	743.843	745.193	746.541	747.887	749.230	750.571	751.909	753.245
10	754.579	755.910	757.239	758.565	759.890	761.212	762.531	763.849	765.164	766.477
20	767.787	769.096	770.402	771.706	773.008	774.307	775.604	776.900	778.193	779.484
30	780.772	782.059	783.344	784.626	785.907	787.185	788.461	789.735	791.007	792.277
40	793.545	794.811	796.075	797.337	798.597	799.855	801.111	802.365	803.617	804.867
50	806.116									
Speed of sound, knots										
-60	568.917	580.804	579.495	578.183	576.869	575.551	574.231	572.907	571.580	570.250
-50	582.109	593.732	592.451	591.169	589.883	588.594	587.303	586.009	584.712	583.412
-40	595.009	606.384	605.131	603.875	602.616	601.355	600.091	598.825	597.555	596.283
-30	607.635	618.778	617.549	616.319	615.086	613.850	612.612	611.371	610.128	608.883
-20	620.003	630.928	629.723	628.516	627.307	626.096	624.882	623.666	622.447	621.227
-10	632.130	642.848	641.666	640.482	639.296	638.107	636.916	635.723	634.527	633.330
0	644.028	645.206	646.382	647.555	648.727	649.896	651.063	652.228	653.391	654.552
10	655.711	656.868	658.022	659.175	660.326	661.475	662.621	663.766	664.909	666.050
20	667.189	668.326	669.461	670.594	671.725	672.854	673.982	675.107	676.231	677.353
30	678.473	679.591	680.707	681.821	682.934	684.045	685.154	686.261	687.366	688.470
40	689.572	690.672	691.770	692.867	693.962	695.055	696.147	697.236	698.324	699.411
50	700.495									

TABLE X.- TEMPERATURE *t* IN DEGREES FAHRENHEIT AND CENTIGRADE AND SPEED OF SOUND *a* IN MILES PER HOUR AND KNOTS FOR VALUES OF PRESSURE ALTITUDE *H* IN GEOPOTENTIAL FEET

Pressure altitude, <i>H</i> , geopotential ft	Temp., <i>t</i> , °F	Temp., <i>t</i> , °C	Speed of sound, <i>a</i> , mph	Speed of sound, <i>a</i> , knots
0	+59.000	+15.000	761.212	661.475
500	+57.217	+14.009	759.902	660.357
1,000	+55.434	+13.019	758.590	659.197
1,500	+53.651	+12.028	757.276	658.055
2,000	+51.868	+11.038	755.960	656.911
2,500	+50.085	+10.047	754.642	655.766
3,000	+48.302	+9.057	753.321	654.618
3,500	+46.518	+8.066	751.997	653.467
4,000	+44.735	+7.075	750.671	652.315
4,500	+42.952	+6.084	749.343	651.161
5,000	+41.169	+5.094	748.013	650.006
5,500	+39.386	+4.103	746.680	648.848
6,000	+37.603	+3.113	745.345	647.687
6,500	+35.820	+2.122	744.013	646.525
7,000	+34.037	+1.132	742.668	645.351
7,500	+32.254	+0.141	741.326	644.195
8,000	+30.471	-0.849	739.981	643.026
8,500	+28.688	-1.840	738.634	641.856
9,000	+26.905	-2.831	737.285	640.683
9,500	+25.121	-3.822	735.932	639.507
10,000	+23.338	-4.812	734.577	638.330
10,500	+21.555	-5.803	733.220	637.151
11,000	+19.772	-6.793	731.861	635.970
11,500	+17.989	-7.784	730.509	634.786
12,000	+16.206	-8.774	729.154	633.600
12,500	+14.423	-9.765	727.767	632.412
13,000	+12.640	-10.756	726.397	631.222
13,500	+10.857	-11.746	725.025	630.029
14,000	+9.074	-12.737	723.650	628.835
14,500	+7.291	-13.727	722.272	627.637
15,000	+5.508	-14.718	720.892	626.438
15,500	+3.725	-15.708	719.509	625.236
16,000	+1.941	-16.699	718.123	624.032
16,500	+0.158	-17.690	716.735	622.825
17,000	-1.625	-18.681	715.344	621.617
17,500	-3.408	-19.671	713.950	620.406
18,000	-5.191	-20.662	712.554	619.192
18,500	-6.974	-21.652	711.155	617.977
19,000	-8.757	-22.643	709.753	616.759
19,500	-10.540	-23.633	708.349	615.538
20,000	-12.323	-24.624	706.941	614.315
20,500	-14.106	-25.614	705.531	613.090
21,000	-15.889	-26.605	704.118	611.862
21,500	-17.672	-27.596	702.702	610.631
22,000	-19.456	-28.587	701.283	609.398
22,500	-21.239	-29.577	699.861	608.163
23,000	-23.022	-30.568	698.437	606.925
23,500	-24.805	-31.558	697.009	605.684
24,000	-26.588	-32.549	695.579	604.441
24,500	-28.371	-33.539	694.146	603.196
25,000	-30.154	-34.530	692.709	601.948
25,500	-31.937	-35.521	691.270	600.697
26,000	-33.720	-36.511	689.828	599.444
26,500	-35.503	-37.502	688.383	598.188
27,000	-37.286	-38.492	686.934	596.930
27,500	-39.069	-39.483	685.483	595.668

Pressure altitude, <i>H</i> , geopotential ft	Temp., <i>t</i> , °F	Temp., <i>t</i> , °C	Speed of sound, <i>a</i> , mph	Speed of sound, <i>a</i> , knots
28,000	-40.852	-40.473	684.029	594.405
28,500	-42.636	-41.464	682.570	593.157
29,000	-44.419	-42.455	681.110	591.868
29,500	-46.202	-43.446	679.646	590.596
30,000	-47.985	-44.436	678.179	589.321
30,500	-49.768	-45.427	676.709	588.044
31,000	-51.551	-46.417	675.236	586.764
31,500	-53.334	-47.408	673.759	585.481
32,000	-55.117	-48.398	672.279	584.195
32,500	-56.900	-49.389	670.796	582.906
33,000	-58.683	-50.379	669.310	581.614
33,500	-60.466	-51.370	667.820	580.320
34,000	-62.249	-52.361	666.327	579.022
34,500	-64.033	-53.352	664.830	577.721
35,000	-65.816	-54.342	663.330	576.418
35,500	-67.599	-55.333	661.827	575.112
36,000	-69.382	-56.323	660.321	573.803
36,500 thru 82,000	-69.700	-56.500	660.052	573.569
82,500	-68.912	-56.062	660.718	574.148
83,000	-68.089	-55.605	661.414	574.753
83,500	-67.266	-55.148	662.108	575.356
84,000	-66.443	-54.691	662.802	575.959
84,500	-65.620	-54.233	663.495	576.562
85,000	-64.797	-53.776	664.188	577.163
85,500	-63.974	-53.319	664.880	577.765
86,000	-63.151	-52.862	665.571	578.365
86,500	-62.328	-52.404	666.261	578.965
87,000	-61.505	-51.947	666.951	579.564
87,500	-60.682	-51.490	667.640	580.163
88,000	-59.859	-51.033	668.328	580.761
88,500	-59.036	-50.576	669.015	581.358
89,000	-58.213	-50.118	669.702	581.955
89,500	-57.390	-49.661	670.388	582.551
90,000	-56.567	-49.204	671.073	583.147
90,500	-55.744	-48.747	671.758	583.742
91,000	-54.921	-48.289	672.442	584.336
91,500	-54.098	-47.832	673.125	584.930
92,000	-53.275	-47.375	673.808	585.523
92,500	-52.452	-46.918	674.490	586.116
93,000	-51.629	-46.461	675.171	586.708
93,500	-50.806	-46.003	675.852	587.299
94,000	-49.984	-45.547	676.531	587.889
94,500	-49.161	-45.089	677.210	588.479
95,000	-48.338	-44.632	677.888	589.069
95,500	-47.515	-44.175	678.566	589.658
96,000	-46.692	-43.718	679.243	590.246
96,500	-45.869	-43.261	679.920	590.834
97,000	-45.046	-42.803	680.595	591.421
97,500	-44.223	-42.346	681.270	592.008
98,000	-43.400	-41.889	681.945	592.594
98,500	-42.577	-41.432	682.619	593.179
99,000	-41.754	-40.974	683.292	593.764
99,500	-40.931	-40.517	683.964	594.349
100,000	-40.108	-40.060	684.636	594.932

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TABLE XI.- TRUE AIRSPEED V IN KNOTS FOR VALUES OF CALIBRATED AIRSPEED Vc IN KNOTS

AND VALUES OF PRESSURE ALTITUDE H IN GEOPOTENTIAL FEET

[Computation of V based on standard temperature at each altitude]

H, geopotential ft	Vc, knots	100	200	300	400	500	600	700	800	900	1,000
0	0	100.0	200.0	300.0	400.0	500.0	600.0	700.0	800.0	900.0	1000.0
5,000	5,000	107.7	215.0	321.6	427.4	532.2	635.8	740.3	847.3	955.2	1063.7
10,000	10,000	116.2	231.6	345.4	457.2	566.8	674.5	785.0	900.5	1017.8	1135.7
15,000	15,000	125.8	250.0	371.5	489.4	603.8	716.3	835.2	960.9	1088.9	1217.5
20,000	20,000	137.2	270.5	400.1	524.4	643.4	763.0	892.4	1029.8	1170.0	1310.8
25,000	25,000	148.7	293.4	431.5	562.0	686.6	816.2	958.0	1109.1	1263.1	1417.7
30,000	30,000	162.4	318.9	465.9	602.6	735.4	877.5	1033.8	1200.6	1370.3	1540.7
35,000	35,000	178.0	347.4	503.6	646.9	791.5	948.7	1121.9	1306.6	1494.4	1682.7
40,000	40,000	199.1	385.6	553.7	708.9	871.5	1049.3	1245.3	1454.1	1666.2	1878.4
45,000	45,000	223.6	429.1	610.0	782.4	967.0	1169.2	1392.1	1628.9	1869.3	
50,000	50,000	251.0	476.4	671.4	865.7	1075.6	1305.6	1558.6	1827.2	2099.4	
55,000	55,000	281.3	527.3	740.2	960.3	1199.2	1460.4	1747.4	2051.7		
60,000	60,000	314.9	581.8	817.9	1068.0	1339.6	1636.2	1961.4	2305.7		
65,000	65,000	351.8	640.4	906.0	1190.5	1499.0	1835.4	2203.6			
70,000	70,000	392.3	705.4	1006.2	1329.8	1679.9	2060.9	2477.6			
75,000	75,000	436.3	778.3	1120.2	1488.0	1885.0	2316.3				
80,000	80,000	484.3	861.2	1249.9	1667.3	2117.1	2604.9				
85,000	85,000	539.0	960.9	1405.3	1881.6	2393.6					
90,000	90,000	599.2	1076.9	1585.8	2129.4	2713.0					
95,000	95,000	663.6	1206.8	1787.3	2405.8						
100,000	100,000	733.8	1351.9	2011.6	2713.1						

